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Jinhang Wu , Ching Tai Ng, Han Fang

**Internal damages detection for structural timber members using low-frequency anti-symmetric guided wave**

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30 **Internal Damages Detection for Structural Timber Members Using Low-frequency**  
31 **Anti-symmetric Guided Wave**

32  
33 Jinhang Wu, Ching Tai Ng\*, Han Fang

34 School of Civil, Environmental & Mining Engineering, The University of Adelaide, SA  
35 5005, Australia

36  
37 **Abstract**

38 Structural timber is one of the commonly used construction materials. Timber can contain  
39 natural internal defect such as knot or natural decay due to its anatomical complexity. Moreover,  
40 internal damage or stiffness reduction can also be induced by environmental and biological  
41 factors such as weathering and termite attacks. This hidden internal damage increases the  
42 difficulty of damage detection using conventional non-destructive testing (NDT) methods.  
43 Ultrasonic guided wave (GW) damage detection technique is one of the promising damage  
44 detection techniques, which can be employed to achieve an effective and robust damage  
45 inspection in timber. However, limited attention has been paid to the use of GW for damage  
46 detection in timber, due to the material anisotropy and inhomogeneity. This paper assesses the  
47 capability of GW in detecting different sizes of the internal damages in a structural red oak  
48 timber using the fundamental anti-symmetric mode ( $A_0$ ) of GW. Measured GW signals in  
49 forward and backward scattering directions are used to calculate the reflection and transmission  
50 ratios for different sizes of internal damages. A series of comprehensive experimental and  
51 numerical parametric studies are carried out using three-dimensional (3D) finite element (FE)  
52 simulations. Good agreement is obtained between numerical and experimental results. The  
53 experimentally verified FE model is utilized to further investigate the wave reflection and  
54 transmission phenomena from different characteristics of internal damages, such as different  
55 lengths, widths, thicknesses, and through thickness locations. The outcomes of this study  
56 demonstrate the robustness of GW technique in detecting conspicuous internal damages in  
57 structural timber. It demonstrates the feasibility of quantitative assessments of internal damage  
58 in timber using  $A_0$  GW.

59  
60 **Keywords:** Timber; damage assessment; internal defect; Lamb wave; finite element; scattering.

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\* Corresponding author email: [alex.ng@adelaide.edu.au](mailto:alex.ng@adelaide.edu.au)

## 61 **1. Introduction**

### 62 **1.1. Backgrounds**

63 Timber has been widely utilised in civil construction due to its sustainable and renewable  
64 nature [1, 2]. Recent developments in non-destructive testing (NDT) and structural health  
65 monitoring (SHM) techniques have enabled a higher standard for damage detection and  
66 structural integrity monitoring in structural members made by timber [3-5]. Visual inspection  
67 is implemented for timber damage evaluation, but it is incapable of identifying the internal  
68 decay or invisible damage [6]. Timber is a natural grow material, it can contain internal defects  
69 such as knot, cracks or natural decay. Physical process on either standing trees or onboard  
70 timber members can also produce internal defects [6, 7]. Consequently, internal defects usually  
71 exist and can affect the structural performance of timber structures. Apart from the natural  
72 defects, internal damages can also be induced by working and environmental conditions. This  
73 involves cracks generated from the mechanical loadings, interior deterioration from high  
74 moisture content (MC) [12] and reduction of bearing capacity from ultraviolet (UV) aging [13].  
75 Moreover, timber structures, especially historical constructions [8], are susceptible to insect  
76 attack due to its organic and biomass nature. The internal flaws can be generated by termites  
77 and beetles with only small surface entry larvae boreholes, which are hard to be detected by  
78 visual inspection [9, 10]. The reduction in timber material for a structural member due to the  
79 internal damage can reduce the stiffness of the structure. Mori et al [11] measured and reported  
80 a noteworthy diminution in timber Young's modulus and bending strength from artificial holes  
81 on timber.

82 To minimize the risk of failures due to deterioration of structural performance of timber  
83 and ensure the structural safety, an effective NDT damage detection method of critical damage  
84 at its early-stage is essential. Compared to metallic materials, such as aluminium and steel,  
85 mechanical behaviour derivations of timber are more complicated due to its anisotropy,  
86 inhomogeneity and presence of natural defects. Hence, different damage detection techniques  
87 for timber have been developed and investigated in the literatures and they are described in the  
88 following sections.

89

## 90 **1.2. Traditional timber damage detection methods**

91 Traditional semi-destructive testing (SDT) methods, such as Resistograph [14, 15] measures  
92 the resistance or properties variation of a timber section by means of electric drilling. However,  
93 destructive testing is prohibited for most of the modern or preserved historical construction.  
94 NDT methods allow evaluating the condition of structures without inducing any damages.  
95 Traditional timber damage detection methods have acoustic emissions, vibration-based  
96 methods, sounding and conventional ultrasonic methods, such as C-Scan. The effectiveness of  
97 a damage detection method to detect the damages depends not only its sensitivity to damage,  
98 but also the operational feasibility considering the realistic environmental conditions.

99 Acoustic emissions [16] is a passive monitoring approach, through which the elastic wave  
100 is generated by a sudden redistribution of stress from localized damages rather an external  
101 excitation source. Consequently, the location of acoustic energy source can be determined.  
102 Vibration based method [17, 18] primarily examines dynamic responses, such as modal  
103 frequencies and modeshapes of structures to determine changes in global stiffness of structures  
104 [19]. This method can detect the existence of large discontinuities or damages in timber  
105 members. However, compared to conventional ultrasonic method, vibration-based method is  
106 less sensitive to small size of damages and incapable of accurately identifying the location of  
107 these small damages [19]. Sounding techniques were utilized to measure the embedded length  
108 or examine the health stage of timber utility pole [20]. However, the sonic wave is usually  
109 created by means of by impact hammer and belongs to broadband signal. Compared to  
110 narrowband signal, broadband excitation signal has higher attenuation and is more dispersive,  
111 which is difficult to analyse the signals. On the other hand, the conventional ultrasonic method,  
112 such as C-Scan, was also commonly used for damage detection of structural timbers [21]. This  
113 method relies on ultrasonic device to generate bulk wave, which was used in detecting small  
114 holes [22, 23] and delamination [24] in timber. In the literature, different studies showed it has  
115 adequate sensitivity to the damages. However, conventional bulk wave method has limited  
116 inspection area: only a small scanning area covered by the ultrasonic device can be inspected.  
117 In summary, these techniques are incapable to effectively detect relatively small-size damages.  
118 Therefore, a robust method to detect early-stage inconspicuous defects in timber is essential.

119

### 120 1.3. Damage detection of timber structures using guided wave

121 Guided wave (GW) technique can provide large area inspection and has high sensitivity to  
122 different types of damage, thus it has the potential to fulfil the requirements of damage  
123 detection for timber. This technique has been widely applied in concrete, metallic and  
124 composites materials in the literature [25, 26]. However, limited attention has been paid on  
125 damage detection of timber using GW.

126 In the literature, most of the studies focused on GW and timber were aimed at identifying  
127 properties of timber material. Dahmen et al [27] determined nine anisotropic constants of  
128 timber using a method combining GWs and bulk waves. The results showed that the  
129 computational cost to include all the timber anisotropies is very high. A number of studies  
130 indicate that GW is capable of detecting the changes in material properties brought by the  
131 environmental factors such as the changes in MCs [28, 29] and UV aging [13]. Fathi et al [28]  
132 recently measured the elastic properties of timber under different MC using GW propagation  
133 method. They showed that the elastic properties determined using GW are in good agreement  
134 with the results obtained from a three-points-bending test. GW has also been recently used to  
135 measure the moisture-dependent viscoelastic properties of timber, such as shear storage, shear  
136 loss and loss factor [29]. These studies indicate GW can provide a more accurate estimation of  
137 the material properties under the changes of MC and UV aging compared to the conventional  
138 bulk wave methods.

139 However, there were limited studies focusing on damage detection of timber members  
140 using GW. One of the main real applications of GW technique in timber is to measure the  
141 embedded length [3, 30] or to assess the health conditions of timber utility poles [20, 31].  
142 Dackermann et al [20] proposed a new method for monitoring the structural health of timber  
143 utility poles. The method combines the machine learning algorithms with GW technique using  
144 a multi-sensor system. El Najjar et al [3] assessed the embedded length and the damage in the  
145 embedded portions of the timber poles using GW propagation method. They concluded that  
146 the wave energy leakage to the surrounding soil is minimal. Numerical studies were also been  
147 performed to simulate the wave propagation on the timber poles. Studies showed that timber  
148 can be assumed as transversely isotropic material in modelling wave propagating on a timber  
149 pole [32, 33]. This is because the stiffness in fibre direction is much greater than the stiffness  
150 in the other two directions direction [34]. This assumption was proven to be valid when treating  
151 the timber pole as one-dimensional (1D) waveguide. Zhang et al [35] also conducted a

152 preliminary study on damage detection of timber using a piezoelectric transducer (PZT). The  
153 wavelet packet energy of GW is calculated for cracks with different lengths.

154 GW technique has the potential to enable a robust damage inspection in timber, however,  
155 limited studies were presented in the literature. Past studies were mainly focused on using GW  
156 to measure the timber properties, and only few studies focused on timber damage detection.  
157 Therefore, development of GW technique in timber damage detection, typically those  
158 inconspicuous internal damages from interior deterioration or insect attack, are remained  
159 critical. To the best of author's knowledge, limited studies have provided quantitative  
160 assessments on internal damages in structural timber.

161 This paper aims to provide a comprehensive wave scattering analysis on the internal  
162 damage in structural timber member using fundamental anti-symmetric mode ( $A_0$ ) GW. To  
163 assess the sensitivity of  $A_0$  GW to the timber internal damage, different sizes of internal damage  
164 have been created experimentally on a structural timber member using a rotary tool. Wave  
165 reflection and transmission ratios are obtained before and after the internal damages. In order  
166 to have a visual understanding and gain insight into the interaction between the internal damage  
167 and the GW, three-dimensional (3D) finite element (FE) simulations are used to model the  
168 internal damage. A series of parametric studies using experimentally verified FE model are  
169 performed to investigate the wave reflection and transmission ratios by varying the length,  
170 width, thickness and location of the internal damages.

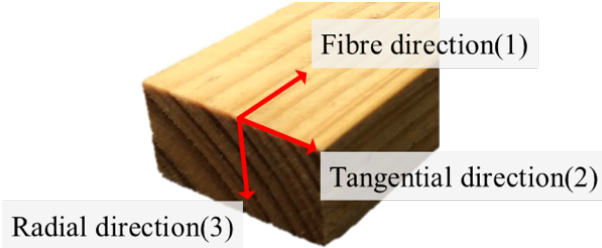
171 The arrangement of the paper is as follows. Section 2 presents the theory of GW  
172 propagation in the transversely isotropic timber. Section 3 describes the details of the 3D FE  
173 model. Section 4 describes the setup and the procedures in the experiment to create different  
174 sizes of internal damages. The reflection and transmission ratios obtained from numerical and  
175 experimental results, along with the numerical case studies using the validated FE, are  
176 discussed in Section 5. The limitations of the present study are discussed in Section 6. Finally,  
177 a conclusion is provided in Section 7.

178

## 179 **2. Governing equations for GW propagating in transversely isotropic material**

180 This section presents the governing equations of the GW propagation in timber. Plane wave  
181 assumption is used in the analytical solution. Timber is an orthotropic material with elastic  
182 properties being different along three principal axes as shown in FIG. 1. Fibre direction is  
183 labelled as 1 while tangential and radial direction are labelled as 2 and 3, respectively. The

184 propagation direction of the plane wave is defined to be consistent to the fibre direction 1. Due  
 185 to the presence of unidirectional fibres, elastic modulus presents large differences between fibre  
 186 direction and radial direction while only minor difference between radial and tangential  
 187 direction [32]. Therefore, timber can be assumed to be transversely isotropic material [32, 34].  
 188



189  
 190 FIG. 1 Three principal axes used in the timber modelling  
 191

192 Governing equations for GWs propagating in an orthotropic plate is given in [36]. For plate  
 193 wave propagation in a transversely isotropic plate, governing equations can be derived by  
 194 simply substituting the restrictions on elastic constants to the existing solutions. The  
 195 compliance matrix of transversely isotropic material can be expressed by taking the inverse of  
 196 the stiffness matrix as shown in Eq. (1a), where  $\varepsilon$  and  $\sigma$  are strain tensor and stress tensor,  
 197 respectively. Five independent variables are required to compute the matrix and they are  
 198  $E_1, E_3, \nu_{13}, \nu_{23}$  and  $G_{13}$ , where  $E, \nu$  and  $G$  are young's modulus, Poisson's ratio and shear  
 199 modulus in the given direction, correspondingly.  $\nu_{ij}$  and  $\nu_{ji}$  are related by  $\frac{\nu_{ij}}{E_i} = \frac{\nu_{ji}}{E_j}$ . Due to  
 200 directions 2 and 3 have the same material elasticity, a symmetric plane is defined for direction  
 201 2-3. Therefore, transverse isotropy has  $E_2 = E_3, \nu_{21} = \nu_{31}$  and  $G_{12} = G_{13}$ , which yields the  
 202 following simplified symmetric matrix.

$$203 \quad \varepsilon_{kl} = c_{ijkl}^{-1} \sigma_{ij} = \begin{pmatrix} \frac{1}{E_1} & -\frac{\nu_{13}}{E_1} & -\frac{\nu_{13}}{E_1} & 0 & 0 & 0 \\ -\frac{\nu_{13}}{E_1} & \frac{1}{E_3} & -\frac{\nu_{32}}{E_3} & 0 & 0 & 0 \\ -\frac{\nu_{13}}{E_1} & -\frac{\nu_{32}}{E_3} & \frac{1}{E_3} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{G_{23}} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{G_{13}} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{G_{13}} \end{pmatrix}, \quad i, j, k, l = 1, 2, 3 \quad (1a)$$



204 
$$G_{23} = \frac{E_3}{2(1 + \nu_{23})} \quad (1b)$$

205 Substituting the stress-strain relationship in Eq. (1) into equations of motion in Eq. (2), coupled  
 206 displacement equations in three directions can be obtained and shown in Eq. (3) [36]. The  
 207 propagation direction of the plane wave is defined to be consistent with the fibre direction 1,  
 208 whereas direction 3 is the thickness direction of the timber.

209 
$$\frac{\partial \sigma_{ij}}{\partial x_j} = \rho \frac{\partial^2 u_i}{\partial t^2} \quad (2)$$

210 
$$u_j = U_j e^{i\xi(x_1 + \alpha x_3 - ct)}, \quad j = 1, 2, 3 \quad (3)$$

211 where  $u_i$  is the displacement component and  $U_j$  displacement amplitude of  $u_j$ .  $\xi$  is  
 212 wavenumber and  $c$  is phase velocity.  $\alpha$  is an unknown representing the ratio of  $x_3$  to  $x_1$   
 213 wavenumbers. Eq. (3) leads to

214 
$$K_{mn}(\alpha)U_n = 0, \quad m, n = 1, 2, 3 \quad (4)$$

215 where the coefficient  $K_{mn}(\alpha)$  is a symmetric matrix, such that  $K_{mn} = K_{nm}$ . For the  
 216 transversely isotropic case, coefficient  $K_{mn}(\alpha)$  are:

217 
$$\begin{aligned} K_{11} &= C_{11} - \rho c^2 + C_{55}\alpha^2 \\ K_{22} &= C_{55} - \rho c^2 + C_{44}\alpha^2 \\ K_{33} &= C_{55} - \rho c^2 + C_{33}\alpha^2 \\ K_{23} &= 0 \\ K_{13} &= (C_{13} + C_{55})\alpha \\ K_{12} &= 0 \end{aligned} \quad (5)$$

218 where contracted notions are defined and used in Eq. (5) to replace  $c_{ijkl}$  in Eq. (1a) with  $C_{ab}$   
 219 following the order of stress tensor index, consequently: 11 = 1, 22 = 2, 33 = 3, 23 = 4, 13 = 5  
 220 and 12 = 6. For consistency of the solutions,  $C_{11}$ ,  $C_{33}$ ,  $C_{44}$ ,  $C_{55}$  and  $C_{13}$  are chosen to express  
 221 all coefficient. According to Nayfeh et al [36], shear horizontal mode (SH) are uncoupled  
 222 mathematically from fundamental symmetric mode (S<sub>0</sub>) and A<sub>0</sub> if wave propagates along  
 223 principle axis in a transversely isotropic plate. Equating determinant of Eq. (4) to zero gives a  
 224 sixth order polynomial [36]:

225 
$$\alpha^6 + B_1\alpha^4 + B_2\alpha^2 + B_3 = 0 \quad (6)$$

226 where  $B_1$ ,  $B_2$  and  $B_3$  are coefficients involving material constants and phase velocity. Both  
 227 coefficients and solution for  $\alpha$  are given in Appendix A.

228 Solve Eq. (6), substitute  $\alpha$  into displacement and stress expressions and apply stress free  
 229 conditions at plate boundaries [36], the characteristic equations for the symmetric and  
 230 antisymmetric Lamb wave modes propagate along principal axis in a transversely isotropic

231 plate can be obtained as shown in Eqs. (7)-(9). Eq. 7(a) is for symmetric waves while Eq. 7(b)  
 232 applies to anti-symmetric waves [36].

$$233 \quad \frac{\tan(\gamma\alpha_1)}{\tan(\gamma\alpha_3)} = \frac{D_{11}D_{23}}{D_{13}D_{21}} \quad (7a)$$

$$234 \quad \frac{\tan(\gamma\alpha_1)}{\tan(\gamma\alpha_3)} = \frac{D_{13}D_{21}}{D_{11}D_{23}} \quad (7b)$$

$$235 \quad \gamma = \frac{\xi d}{2} \quad (8)$$

$$236 \quad \begin{aligned} D_{1k} &= (C_{13} + C_{33}\alpha_k W_k) \\ D_{2k} &= C_{55}(\alpha_k + W_k) \\ W_k &= \frac{\rho c^2 - C_{11} - C_{55}\alpha_k^2}{(C_{13} + C_{55})\alpha_k} \end{aligned} \quad (9)$$

237 where  $d$  is the thickness of the plate and  $k = 1,2,3,4,5,6$ . The corresponding roots for  $\alpha$  are  
 238 given in Appendix A. Semi-analytical solutions of Eqs. (7)-(9) are shown in Section 3.1.

239

### 240 3. 3D FE model

241 3D FE models were developed using ABAQUS/Explicit to simulate the GW propagation and  
 242 wave scattering phenomena in timber. Reflection and transmission ratios from different sizes  
 243 of internal damage were obtained. An experiment was performed to validate the FE results.  
 244 The validated FE model was then used to perform a series of parametric studies, and the results  
 245 are given in Section 5.

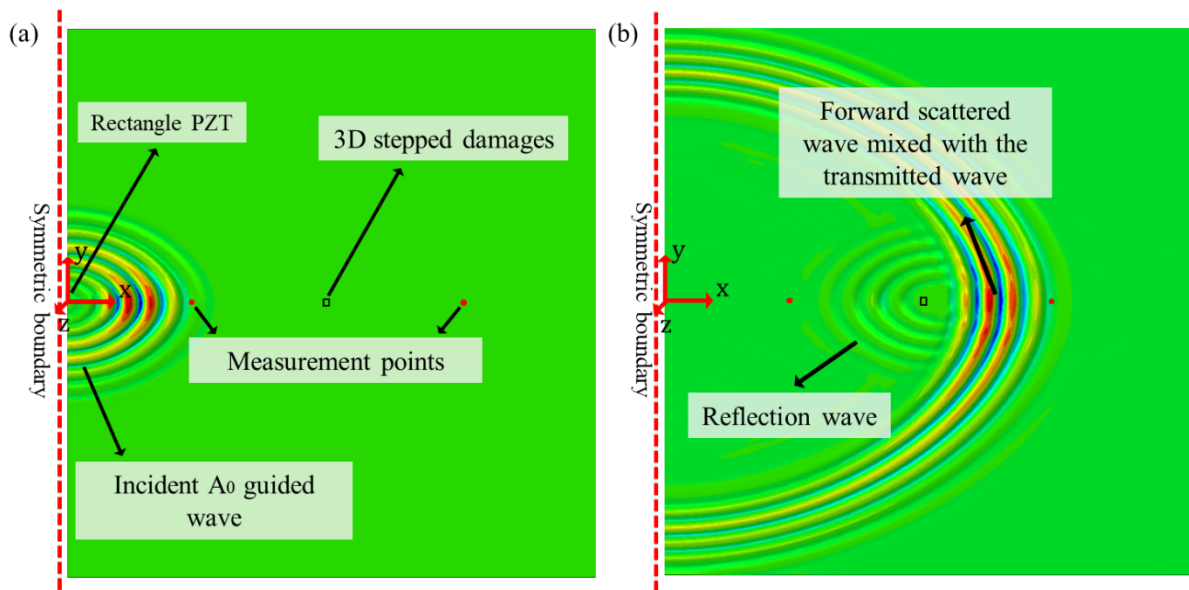
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#### 247 3.1. Model definition

248 A 10 mm thick Tasmanian red oak was modelled in ABAQUS, which is the same as the  
 249 specimen used in the experiment. To avoid unwanted reflections from the edges, the width and  
 250 length of the timber were set to be 800 mm, which are large enough to avoid wave reflections  
 251 from edges. The configuration of the model is shown in FIG. 2(a). A Cartesian coordinate is  
 252 established to describe the locations of measurement points. The coordinate is defined at the  
 253 surface of the timber, and the origin of system is defined at the excitation centre.  $x$  direction is  
 254 aligned with timber fibre direction, which is also the direction of wave propagation. A  
 255 symmetric boundary condition with respect to  $x$  axis was applied. Different sizes of internal  
 256 damages were created at  $x = 290$ mm. To ensure the reflection wave is separated from the  
 257 incident wave, the reflection wave measurement point is at  $x = 100$ mm while the transmission

258 wave measurement point is at equal distance away from the damage with respect to reflection  
 259 scanning point. The timber is modelled using 3D eight-node linear brick elements with reduced  
 260 integration and hourglass control (C3D8R) element. To investigate the precision of using  
 261 reduced integration element, 3D eight-node linear brick elements with incompatible mode, full  
 262 integration, and hourglass control (C3D8I) were also used to model timber. C3D8I not only  
 263 utilises full integration but also has additional degree of freedom, which can also capture  
 264 bending motion. Identical results were obtained from these two models. Since C3D8I is more  
 265 computationally expensive, C3D8R was used in this study.

266



267

268 FIG. 2 Screenshot of out-of-plane displacements in z axis in a 10 mm red oak at different  
 269 time steps: (a) right after incident A<sub>0</sub> wave was generated; (b) interaction of A<sub>0</sub> wave with the  
 270 3D stepped damages.

271

272 Red oak was modelled using transversely isotropic material in this study. The material  
 273 properties are listed in Table 1 and density of the timber was measured to be 647.7 kg/m<sup>3</sup> [34].  
 274 Fibre direction ( $E_{11}$ ) is aligned with x direction as shown FIG. 2. Young's modulus of fibre  
 275 direction ( $E_{11}$ ) is measured experimentally by means of time-of flight using GW, which is 10.5  
 276 GPa, shared a similar value of 8.91 GPa from reference [34].

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281

Table 1 Elastic properties of Tasmanian red oak

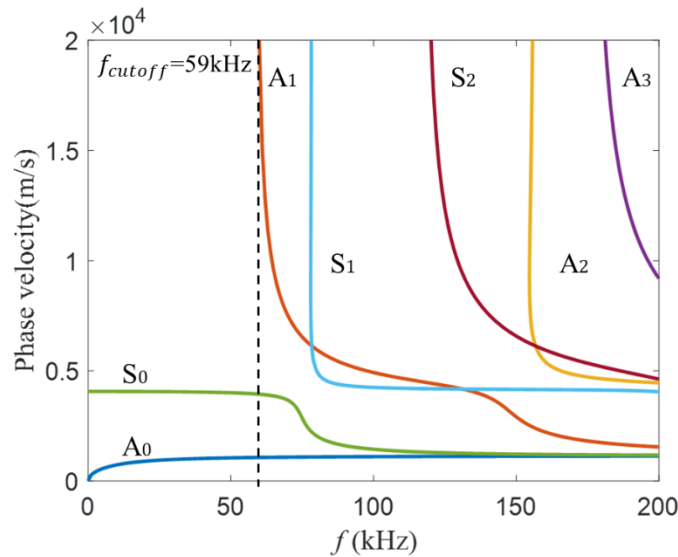
$E_{11}$ (GPa)	$E_{22}$ (GPa)	$E_{33}$ (GPa)	$\nu_{12}$	$\nu_{13}$	$\nu_{23}$	$G_{12}$ (GPa)	$G_{13}$ (GPa)	$G_{23}$ (GPa)
10.50	1.24	1.24	0.05	0.05	0.43	0.89	0.89	0.43

283

284 To simulate the wave attenuation in the model, Rayleigh damping is employed to simulate  
 285 energy dissipation during wave propagation [25]. Attenuation constant  $k_i$  represents the rate  
 286 of energy dissipation along the direction of the wave propagation and it is determined by fitting  
 287 an exponential function to the experimental data for the decrement of signal amplitude versus  
 288 distance [37, 38]. Eq. (10) shows the formula to obtain  $k_i$ , where  $x_1$  and  $x_2$  are the locations of  
 289 measurement points, while  $A_1$  and  $A_2$  are the wave amplitudes, correspondingly.

$$290 \quad \frac{A_1}{A_2} = \exp(k_i(x_2 - x_1)) \quad (10)$$

291 Mass-proportional damping constant ( $\alpha_\omega$ ) and stiffness-proportional constant ( $\beta_\omega$ ) can then be  
 292 computed using  $k_i$ .  $\alpha_\omega = 2k_i c_g$  and  $\beta_\omega = \frac{\alpha_\omega}{\omega^2}$ , where  $c_g$  and  $\omega$  are the group velocity and  
 293 angular central frequency.



294

295 FIG. 3 Dispersion diagram for wave propagation along  $0^\circ$  in  $x$  direction in 10 mm

296

Tasmanian red oak

297

298 The dispersion relationships for Lamb wave propagating along  $0^\circ$  in  $x$  direction in 10 mm  
 299 Tasmanian red oak can be obtained by solving the governing equations Eqs. (7)-(9). The wave  
 300 modes for up to 200 kHz are shown in FIG. 3. As shown in the FIG. 3,  $A_0$  wave has smaller

301 phase velocity and wavelength than  $S_0$  GW under same excitation frequency. Therefore,  $A_0$   
302 GW is more sensitive to smaller damages, and hence, it is selected as the excitation wave mode  
303 in this study. The cut off frequency is 59 kHz. To minimize the complexity in wave analysis of  
304 higher orders and multi-modes, the excitation frequency is chosen below 59kHz. Theoretically,  
305 a higher frequency can have a smaller wavelength, which is more sensitive to smaller size of  
306 damage. However, it is found that 35 kHz can provide the best signal-to-noise ratio for the  
307 experimentally measured signal. Therefore, the excitation signal is selected as a 5-cycle  
308 narrow-band 35 kHz Hann windowed pulse and used for the rest of the study.

309 In the FE simulation, the  $A_0$  wave is generated by applying out-of-plane displacements on  
310 a  $12\text{ mm} \times 6\text{ mm}$  rectangle region, which has the same size and shape as the PZT used in the  
311 experiment. The maximum element size and time increment were recommended to be less than  
312  $\Delta l = \frac{\lambda}{20}$ ;  $\Delta t = \frac{1}{20f}$  to ensure the stability and accuracy of explicit analysis [39], where  $\Delta l$  is  
313 the maximum element size,  $\Delta t$  is the time step,  $\lambda$  is the wavelength of excitation signal, and  $f$   
314 is the central excitation frequency. The wavelength is 28 mm for  $A_0$  GW under selected  
315 excitation frequency 35 kHz. Therefore, the maximum element size is set to be 1 mm to ensures  
316 there are at least 28 elements existed per wavelength. To ensure the out-of-plane displacement  
317 is accurately modelled, the thickness of the element is set to be 1 mm and there are 10 layers  
318 of element in the thickness direction.

319 FIG. 2 (a) shows a snapshot of out-of-plane displacement in FE right after  $A_0$  GW is excited.  
320 As shown in the displacement contour, the energy of incident  $A_0$  wave concentrates along the  
321 fibre direction, which has dominant stiffness. Therefore, the measurement locations are defined  
322 along fibre direction to capture most of the wave energy. FIG. 2 (b) shows the interaction of  
323  $A_0$  with the internal damage. As a result, reflection wave and forward scattered wave are  
324 generated at the damage. It can also be seen that the forward scattered wave is mixed with the  
325 transmitted wave.

326

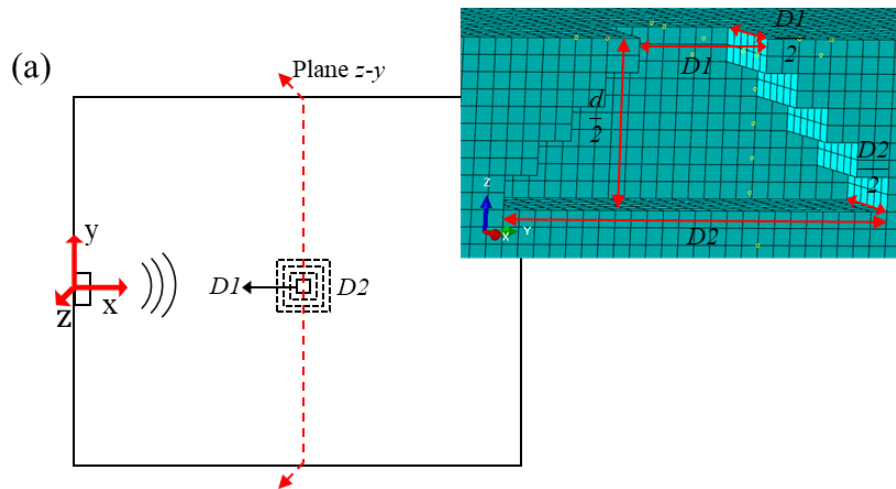
### 327 **3.2. Simulation of internal damages**

328 Internal damages were approximated as 3D semi-stepped damages in the model validation.  
329 This approximation was developed from a 2D stepped notch damage model, which was  
330 simulated to represent corrosion damage in metallic materials [40]. It was found that the  
331 stepped notches provide a realistic approximation to the corrosion damage. Therefore, it was

332 utilized and further enhanced to 3D semi-stepped damages to represent the internal damages in  
 333 the experiment.

334 Internal damages were created experimentally with a small surface entry, whose diameter  
 335 equal to  $D_1$ . The diameter of the damage gradually increases from  $D_1$  at the surface and reaches  
 336 a maximum diameter  $D_2$  at the mid-depth ( $\frac{d}{2}$ ) of the timber, where  $d$  is the thickness of the  
 337 timber. A schematic diagram of a 3D semi-stepped damage is shown in FIG. 4. As it can be  
 338 seen in the  $x$ - $y$  plane view, each layer of the 3D semi-stepped damage composed by a square.  
 339 The diameter of the square damage gradually increases from the top surface ( $D_1$ ) to mid-plane  
 340 ( $D_2$ ). Different internal damage cases were created by enlarging  $D_2$ , while  $D_1$  and depth of the  
 341 damage remained the same in the experiment. However, due to  $D_1$ ,  $D_2$  and depth of the damage  
 342 were fixed, an identical 3D stepped damage with “unit increasing step” is inapplicable. For  
 343 example, when there are  $n$  layers of squares, each layer will possess a thickness of  $d/2n$  mm.  
 344 However, the length and width of the squares in each layer are fixed as  $(D_2 - D_1) / (2n - 2)$  mm.  
 345 Therefore, if  $h/n$  is not equal to  $(D_2 - D_1) / (2n - 2)$ , the shape of each “unit step” becomes  
 346 “semi-stepped”. Denser elements were used to mesh the damage region to achieve good  
 347 elements aspect ratio. Different sizes of 3D semi-stepped damages were validated by the  
 348 experimental results and shown in Section 5.1. The experimentally validated FE model was  
 349 then used to perform a series of numerical parametric studies to obtain the reflection and  
 350 transmission ratios for different internal damage cases in Sections 5.2 and 5.3.

351



352

353

FIG. 4 Schematic diagram for a 3D FE semi-stepped damage

354

#### 355 4. Experiment

356 An experiment was conducted to provide validation of the FE model, and physical  
357 understanding of the GW propagation in the timber and scattering at the internal damage.  
358 Reflection and transmission waves were measured for different sizes of internal damages to  
359 obtain the reflection and transmission ratios. A  $10 \times 90 \times 1000 \text{ mm}^3$  ( $d \times w \times l$ ) red oak  
360 structural timber was used in this study, which has the same thickness as FE model. Red oak is  
361 categorized as hard wood, which possesses a stiffer rigidity compared to most softwoods.  
362 Specimen dimensions are the same as the original product to ensure the practicability of the  
363 experiment. The properties of oak were determined as described in Section 3.1. All  
364 experimental measurements were conducted in an indoor environment on the same day. The  
365 indoor temperature was maintained at  $27^\circ$ , and hence, the moisture content of timber sample is  
366 assumed to remain unchanged. Internal damage generation, and the measurements for  
367 reflection and transmission wave are performed on a selected region of the specimen. The  
368 timber grain of the region is selected to be smooth, and hence, no natural defects or obvious  
369 cracks are presented. Therefore, it is ensured that the received scattering waves are only  
370 generated by the internal damage

371 The  $A_0$  wave was excited by a  $2 \times 12 \times 6 \text{ mm}^3$  ( $d \times w \times l$ ) surface-mounted rectangle PZT.  
372 The PZT was placed at 300 mm away from the right end and at the centre of the specimen. The  
373 locations of reflection and transmission wave scanning points, and the locations of internal  
374 damage are the same as those used in the FE simulation as shown in FIG. 2. The excitation  
375 signal was a 5-cycle narrow-band 35 kHz Hann windowed pulse, generated by a NI PXIE-5122  
376 signal module. Due to the narrow width of the specimen, absorbing clay was attached at four  
377 edges of the timber to absorb unwanted wave reflections from edges. A Kron-Hite 7500  
378 amplifier was used to amplify signal voltage. Signals were measured by a Polytec 1D scanning  
379 laser Doppler vibrometer before and after the wave interaction with the damages [41, 42]. Since  
380 the focus of this study is on  $A_0$  GW, only out-of-plane displacements were measured.  
381 Reflective paints were evenly sprayed on scanning areas to improve light reflections on sample  
382 surface for the measurements. Sampling frequency was set to be 10.24 MHz with 1200 times  
383 samples averaging. A photo and a figure of schematic experimental setup are shown in FIG.  
384 5(a).

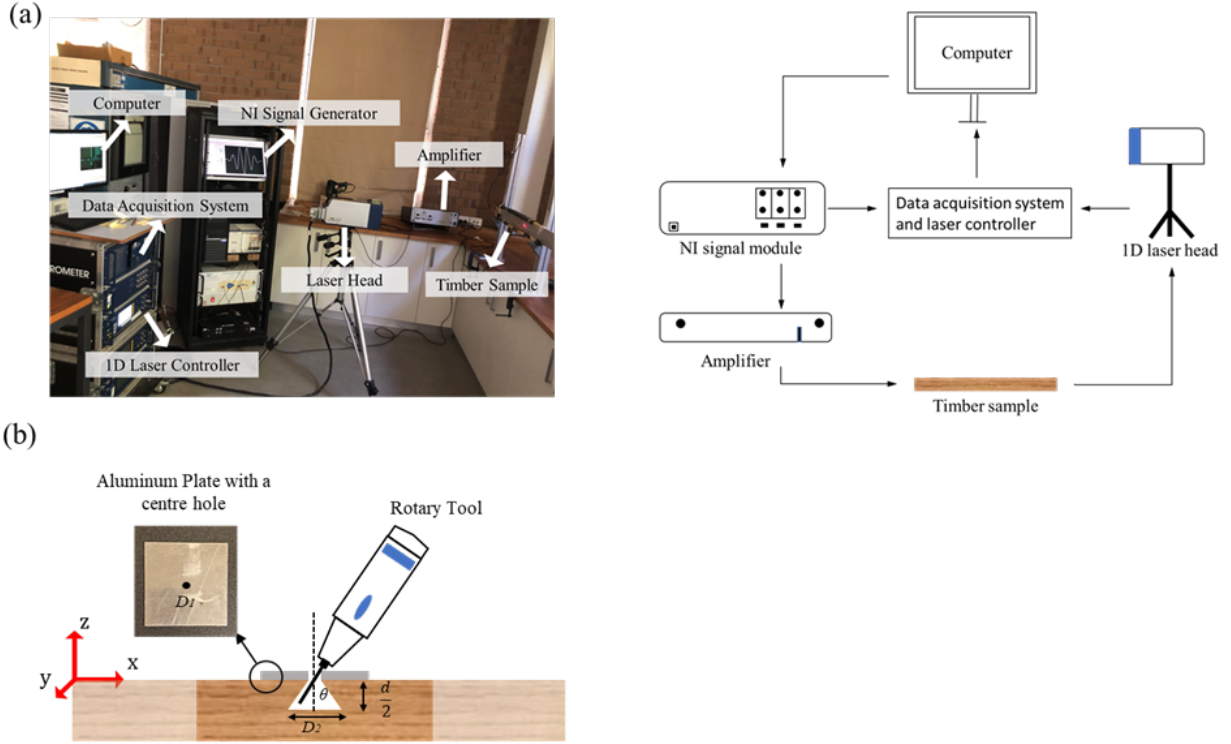


FIG. 5 (a) experimental setup and (b) zoomed view of the sections where internal damages were implemented in experiment

#### 4.1. Experimental validation

Before implementing internal damages on the specimen, the accuracy of the wave simulation in FE was validated by experimental results and the analytical dispersion curve. A 2D discrete Fourier transform (DFT) method was used in this study to validate  $A_0$  GW in both numerical models and experiment results. 2D DFT method transforms the time-domain signals, which acquired from numbers of equally spaced measurement points along wave path, to frequency-wave number domain, so that superimposed wave signal resulting from multi-modes can be separated [43]. Applying 2D DFT to space-time signal  $u(x, t)$  yields the following results [43]:

$$H_{\xi+1, f+1} = \frac{1}{MN} \sum_{n=1}^N \sum_{m=1}^M u_{n,m} e^{-2\pi i (\frac{\xi}{N}(n-1) + \frac{f}{M}(m-1))} \quad (10)$$

where  $f$  and  $\xi$  donate frequency and wavenumber;  $u_{n,m}$  is the space-time signal consisting of  $M$  time samples at  $N$  measurement points.

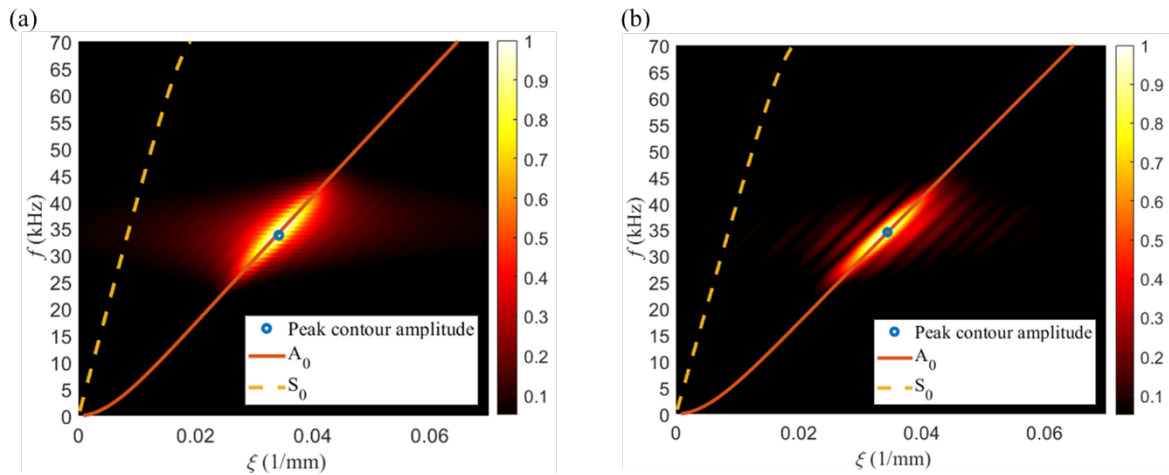
Time-domain signals at 60 evenly spaced points along fibre direction were measured on the intact specimen. The excitation wave was a 5-cycle narrow-band 35 kHz Hann windowed pulse, which is the same as the FE model. The same Cartesian coordinate in FIG. 2 is used to



403 describe the measurement locations. The first measurement point is located at  $x = 100$  mm, the  
 404 rest of the points were distanced by an equal distance of 5 mm. It was experimentally  
 405 determined that the number of measurement points and an equal measurement spaced of 5 mm  
 406 are good enough to produce an acceptable resolution in the contour. In addition, zero paddings  
 407 were used in time-domain and space-domain data to increase resolutions of the plots. Identical  
 408 measurement locations were also defined in the FE model for performing the 2D DFT.

409 The 2D DFT results obtained from FE and experiment are shown in FIG. 6 (a) and FIG. 6  
 410 (b), respectively. A wavenumber-frequency contour is plotted with the analytical dispersion  
 411 curves for the 10 mm thick oak timber under the excitation frequency of 35 kHz. As shown in  
 412 FIG. 6 (a) and (b), there is excellent agreement for the experiment and FE data. The contours  
 413 have good agreement with analytical  $A_0$  dispersion curves. Moreover, contours reach a  
 414 maximum amplitude around 35 kHz, which is the centre frequency of the excitation pulse and  
 415 marked using a blue circle. The results show that  $A_0$  is strongly dominant under the low  
 416 frequency condition (35 kHz), and  $S_0$  GW is not observed from the contour.

417 To capture the wave attenuations with FE, the out-of-plane displacements were also  
 418 measured at the scanning area. By using Eq. (10), Rayleigh damping constants were computed  
 419 as  $k_i=1.965$  Np/m,  $\alpha_\omega = 4677.093$  rad/s and  $\beta_\omega = 9.671 \times 10^{-8}$  s/rad.



420  
 421 FIG. 6. Wavenumber-frequency contour plot and the analytical dispersion curves (solid  
 422 line:  $A_0$ ,  $S_0$ : dashed line) for a 10mm thick red oak, (a) FE results (b) Experimental results by  
 423 35kHz incident wave.  
 424

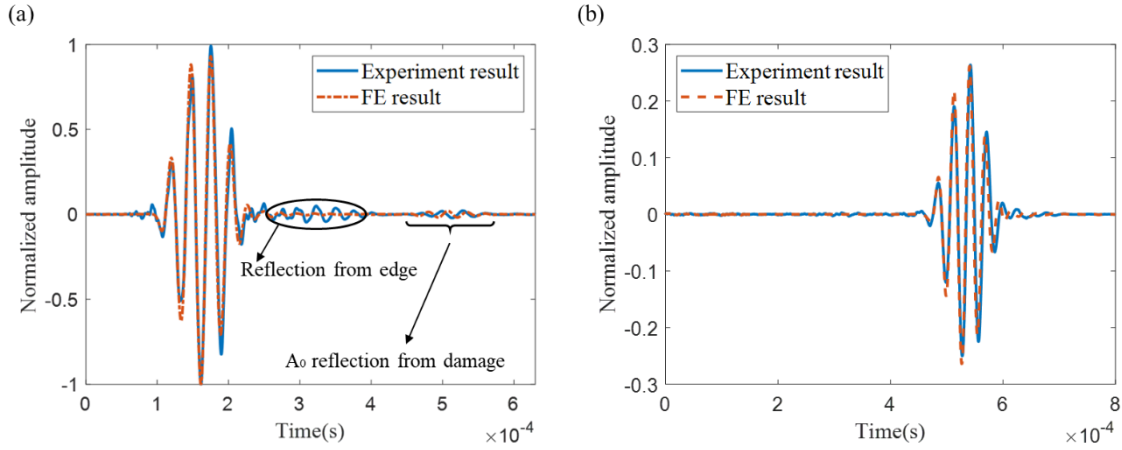
## 425 4.2. Generation of internal damage

426 Five different sizes of internal damages were investigated to obtain the reflection and  
427 transmission wave ratios. The locations of reflection and transmission wave scanning points,  
428 and the locations of internal damage are the same as in those used in the FE simulation as  
429 shown in FIG. 2.

430 FIG. 5 (b) shows a schematic diagram of the internal damage implementations. The  
431 diagram zooms in on the sections where internal damage is implemented. Experimentally, an  
432 internal damage cannot be created without comprising the integrity of the surface of the timber  
433 specimen. Therefore, a small surface hole with diameter ( $D_1 = 6$  mm) is created to allow an  
434 internal access. The diameter of the internal damage increases gradually from the surface ( $D_1$ )  
435 and reaches the maximum diameter  $D_2$  at mid-plane ( $\frac{d}{2} = 5$  mm) of the timber using a rotary  
436 tool. To control the sizes of internal damage diameters, an aluminium plate with a centre hole  
437 (diameter is equal to  $D_1$ ), was temporarily mount on top of the timber surface. The centre hole  
438 on the aluminium plate provides a fixed rotational angle ( $\theta$ ) between rotary head and the  
439 vertical line, which provide a reasonable accuracy in generating the  $D_2$  of the damage. To create  
440 different sizes of  $D_2$ , different sizes of rotary heads were substituted in the rotary tool. Each  
441 internal damage was created by rotating the rotary tool in  $360^\circ$ , which was considered to have  
442 equal damage extent in fibre and tangential direction. Due to the availability of rotary heads  
443 and aforementioned difficulties in control of the sizes  $D_2$ , five discrete sizes of  $D_2$  were created  
444 (7 mm, 9 mm, 12 mm, 14 mm and 16 mm) providing that the depth of the damage and  $D_1$   
445 remain 5 mm and 6 mm, respectively.

446 Time signals were recorded for each damage size at the aforementioned locations. In  
447 addition, a fan was used to clean the accumulative wood dust inside the damage before the next  
448 drilling progression. FIG. 7 (a) and (b) compare the FE results with the experiment time signal  
449 at reflection and transmission measurement locations for the case when  $D_2 = 7$  mm. The time  
450 domain signals at reflection and transmission points are normalized with respect to the incident  
451 wave amplitude at the reflection measurement point. Both figures confirmed that the time of  
452 arrivals and the wave amplitudes in experiments are accurately predicted by FE. Mode  
453 conversion effect could occur through the interaction between  $A_0$  GW and the internal damage,  
454 from which, the in-plane waves could be generated. However, the time domain data was  
455 recorded in out-of-plane direction only due to the focus of this study is on  $A_0$  GW. The  
456 magnitudes of the resulting in-plane waves are very small and have not been observed in the  
457 out-of-plane time domain data in both numerical and experimental data. Minor phase shifts are

458 observed in FIG. 7, and due to following reasons: 1) the presence of local inhomogeneities in  
 459 timber sample and 2) slight misalignments of damage size or location. It is also noticed there  
 460 is a small unabsorbed wave component from the first reflection at sample width in FIG. 7(a).  
 461 The unabsorbed wave component does not interfere with the reflection components from the  
 462 damage as it can be seen that the two components are clearly separated.



463  
 464 FIG. 7 FE and experimental time signal for internal damage size  $D_2 = 7$  mm at (a)  $x =$   
 465 100mm (b)  $x = 480$  mm by 35kHz incident wave  
 466

## 467 5. Results of numerical case studies

468 As described in Sections 3 and 4, reflection and transmission waves before and after the  
 469 internal damages were captured. Three ratios are defined to quantify wave reflection and  
 470 transmission ratios, which are reflection wave ratio  $r_r$ , transmission wave ratio  $r_{tr}$  and forward  
 471 scattered wave ratio  $r_{trb}$ . These wave pockets are shown in FIG. 2. Forward scattered wave is  
 472 retrieved by means of baseline subtraction due to it mixes with the transmission wave.

473 The ratios are calculated based on the same normalization method as before. They are  
 474 normalized by the areas of the incident wave at the reflection point as follow:

$$475 \quad r_r = \frac{A_r}{A_i}, \quad r_{tr} = \frac{A_{tr}}{A_i}, \quad r_{trb} = \frac{A_{tr} - A_{udtr}}{A_i} \quad (10)$$

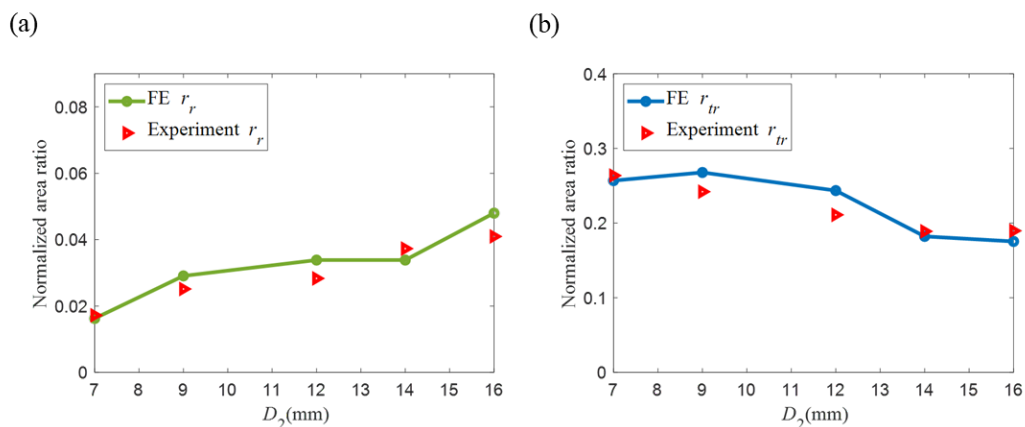
476 where  $A_i, A_r$  and  $A_{tr}$  represent the areas under the incident pulse, reflection pulse, and  
 477 transmission pulse in time-domain signal, respectively. The time-domain signals were  
 478 computed as the absolute values when used to calculate the areas under the signal.  $A_{udtr}$   
 479 represents the areas under transmission wave for the undamaged case. As aforementioned, it is  
 480 used for the purpose of baseline subtraction to obtain the forward scattered wave ratio ( $r_{trb}$ ).

481  $r_r$  and  $r_{tr}$  from the experimental and numerical results are shown in Section 5.1. The  
 482 experimentally validated 3D FE model was then used to perform a series of parametric studies  
 483 to further investigate the reflection and transmission effects from different internal damage  
 484 cases. The wave measurement locations were defined consistently as before, and the wave  
 485 excitation frequency was the same as 35 kHz. Section 5.2 shows the effects of internal damages  
 486 with different dimensions. Section 5.3 shows the effects of the through thickness locations of  
 487 internal damages.

488

### 489 5.1. Experimental results

490  $r_r$  and  $r_{tr}$  obtained from five different sizes of internal damages are shown in FIG. 8(a) and  
 491 (b). As for  $r_{trb}$ , baseline subtraction is impractical to measure in the experiment. It is because  
 492 the specimen needs to be taken off from the laser scanning frame for creating the damage,  
 493 hence, large phase shifts can be induced from a minor change in measurement location or a  
 494 minor delay in wave generation each time. In general, the experimental results exhibit  
 495 consistent trends as the FE results. Both experimental and FE results show that the  $r_r$  increases  
 496 with the sizes of  $D_2$  and reaches a peak at  $D_2 = 16$  mm, while the  $r_{tr}$  presents the opposite trend.  
 497 A small discrepancy between FE and experiment predictions is observed in both figures. This  
 498 is due to minor mismatches of shapes and sizes of the internal damages. From FIG. 8, the size  
 499 of  $D_2$  can be estimated from the reflection and transmission area ratio. Despite the internal  
 500 damages in the reality have irregular shapes, the experimental results still indicate the great  
 501 sensitivity and potentials of GW for detecting timber internal damages.



502

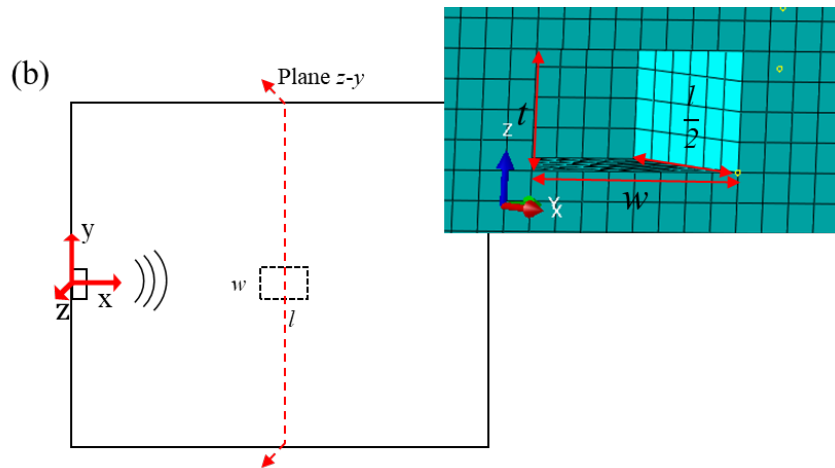
503 FIG. 8 Normalized area ratio for the (a) reflection wave (b) transmission wave resulting

504

from internal damages by 35kHz incident wave

505 **5.2. Effects of internal damages with different dimensions**

506 The experimentally verified FE model was used to further investigate the reflection and  
507 transmission ratios for varying length ( $l$ ), width ( $w$ ), and thickness ( $t$ ) of the internal damage.  
508 As shown in FIG. 9, the internal damages were modelled in an ideal cuboid shape to quantify  
509  $l$ ,  $w$  and  $t$  of the damage. The centroid of the internal damage is located at the mid-plane of the  
510 timber and is axial symmetric in both plane  $x$ - $y$  and plane  $y$ - $z$ .  
511



512

513

FIG. 9 Schematic diagram for FE internal damage with different dimensions

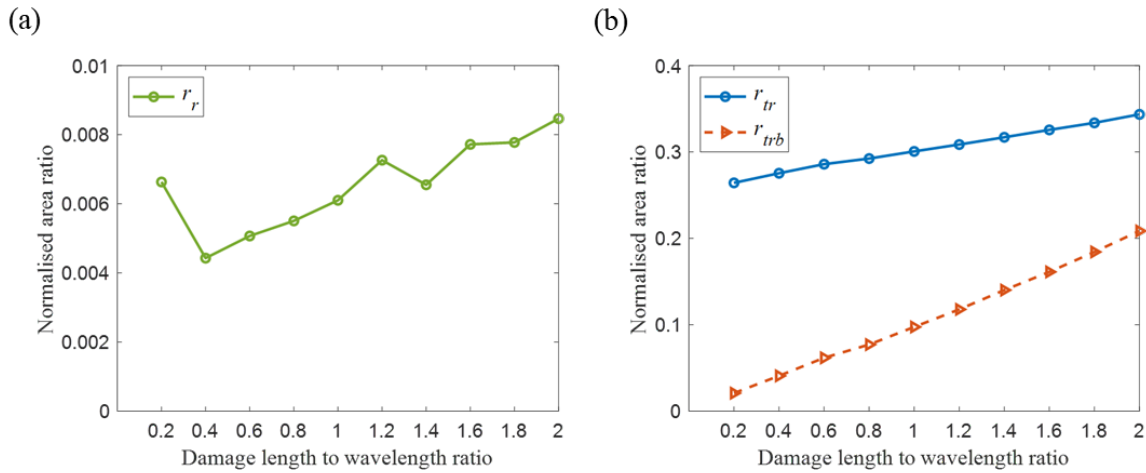
514

515  $l$  and  $w$  of the damage were quantified with respect to wavelength ( $\lambda$ ), while  $t$  of the damage  
516 was quantified with respect to the thickness ( $d = 10$  mm) of the timber. The dimensions of the  
517 damage were initialized by  $0.2\lambda \times 0.2\lambda \times 0.2d$  ( $l \times w \times t$ ). Parametric studies were performed to  
518 vary one of these three parameters  $l$ ,  $w$  and  $t$ , and keep the other parameters unchanged in the  
519 initial stage.  $l$  and  $w$  of the damage were swept from zero to  $2\lambda$  with a step of  $0.2\lambda$  respectively.  
520  $t$  of the damage was swept from zero to a hundred percent of the  $d$  with a step of  $0.1d$ . The  
521 parametric results are presented in the sequence of varying  $l$ ,  $w$  and  $t$  as shown in the following  
522 paragraphs. In addition to  $r_r$  and  $r_{tr}$ ,  $r_{trb}$  is also calculated. This is due to an ideal undamaged  
523 baseline is available from the numerical model.

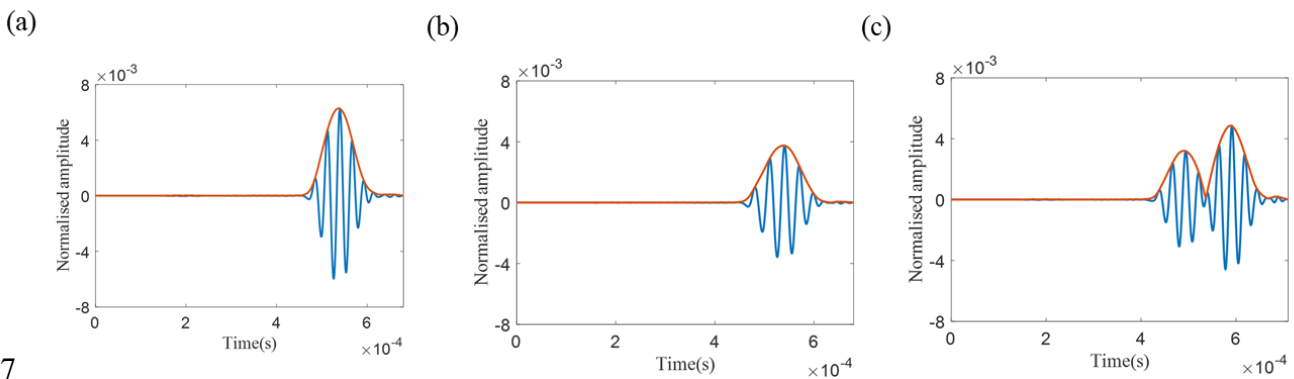
524

525 FIG. 10 shows the results for varying  $l$ . In summary,  $r_{tr}$  and  $r_{trb}$  presented a more intuitive  
526 linear increasing trend compared to the  $r_r$ . This is due to  $r_r$  is affected by the interference  
527 between the first reflection pulse from the start of the damage and second reflection pulse from  
528 the far end of the damage. Two reflection pulses can either interfere constructively and  
destructively when length to wavelength ratio is equal to 0.2 and 0.3, respective, as shown in

529 FIG. 11(a) and FIG. 11(b). A clear separation of two reflection waves can be observed when  
 530 length to wavelength ratio is equal to 1.8 as shown in FIG. 11(c).  
 531



532  
 533 FIG. 10 Normalized area ratio for varying damage length for the (a) reflection wave (b)  
 534 transmission wave (solid line) and forward scattering wave (dashed line) by 35kHz incident  
 535 wave  
 536

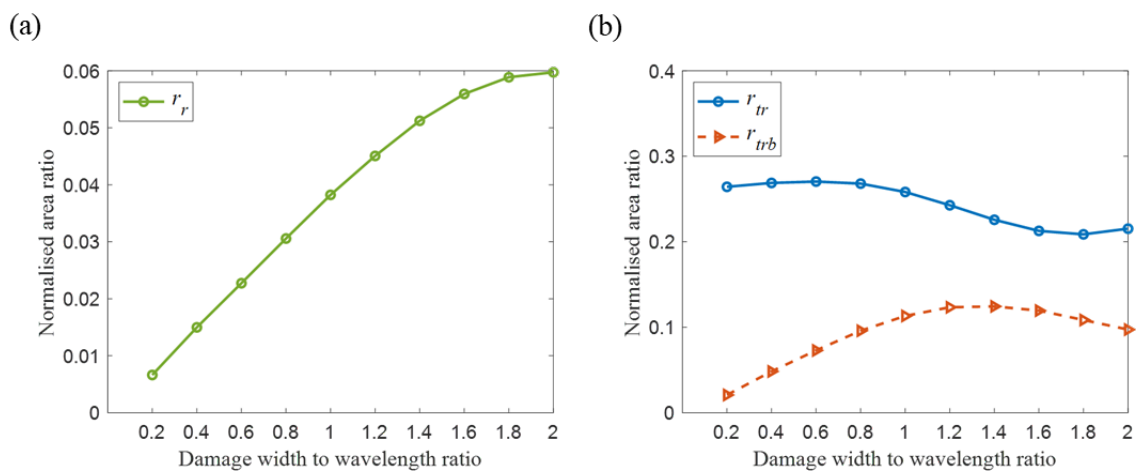


537  
 538 FIG. 11 Time signal of reflection wave at damage length to wavelength ratio equals to  
 539 (a) 0.2, (b) 0.3 and (c) 1.8 by 35kHz incident wave  
 540

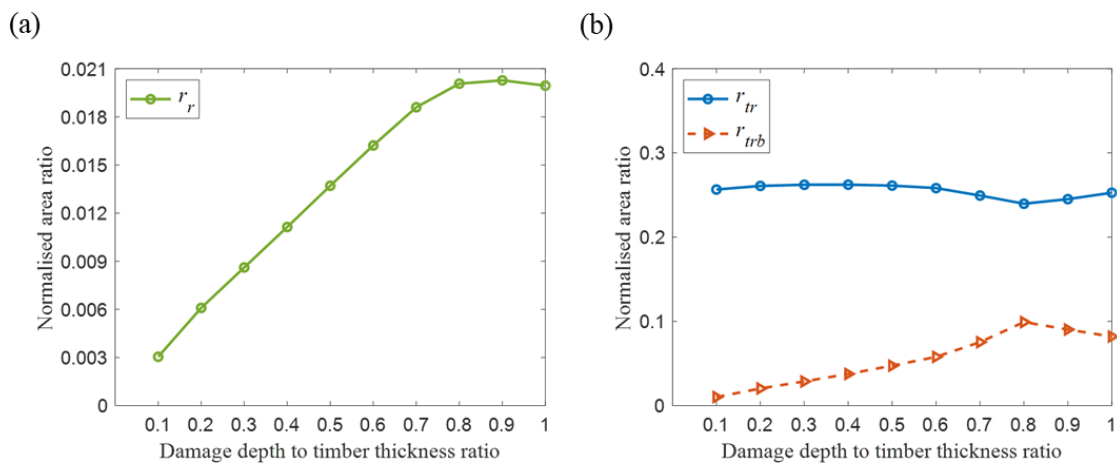
541 FIG. 12 and FIG. 13 show the reflection and transmission results for varying  $w$  and  $t$  of  
 542 the damage. As aforementioned, the  $l$  of the damage was kept to 0.2 wavelength, which  
 543 minimize the destructive interference effects from interference phenomenon between two  
 544 reflection waves as shown in FIG. 11(a). Despite the width and thickness of the damage were  
 545 quantified differently, i.e. width was quantified with respect to wavelength while thickness was  
 546 to the thickness of the timber, the trends of  $r_r$ ,  $r_{tr}$  and  $r_{trb}$  in both cases display almost identical

547 patterns. The results clearly indicate that  $r_r$  increases monotonically with the increasing  
 548 damage width and depth, while  $r_{tr}$  and  $r_{trb}$  experienced a slight fluctuation.

549 In summary,  $r_r$  shows a more intuitive increasing pattern for the increasing size of internal  
 550 damage compared to  $r_{tr}$  and  $r_{trb}$  and therefore more suitable to be used for identifying the  
 551 increasing size of the internal damage. Despite in general,  $r_{tr}$  and  $r_{trb}$  have larger amplitude  
 552 than  $r_r$ , it is found that the patterns of  $r_{tr}$  and  $r_{trb}$  can fluctuate for different damage cases.  
 553 Moreover, the measurement of the transmission wave requires the additional access to the far  
 554 end of the damage, and hence, it is less practical compared to the measurement of the reflection  
 555 wave.



556  
 557 FIG. 12 Normalized area ratio for varying damage width for the (a) reflection wave (b)  
 558 transmission wave (solid line) and forward scattering wave (dashed line) by 35kHz incident  
 559 wave



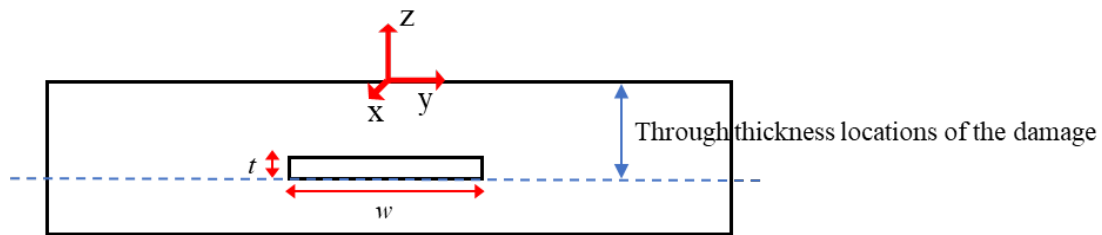
561  
 562 FIG. 13 Normalized area ratio for varying damage thickness for the (a) reflection wave  
 563 (b) transmission wave (solid line) and forward scattering wave (dashed line) by 35kHz  
 564 incident wave

565

### 566 5.3. Effect of internal damages with different thickness locations

567 Section 5.2 shows that results of the reflection and transmission wave for varying  $l$ ,  $w$  and  $t$  are  
568 different. On the other hand, internal damage can exist in any through thickness locations.  
569 Therefore, it is also important to investigate the sensitivity of  $A_0$  GW to the through thickness  
570 locations of the internal damage. Internal damages with different though thickness locations  
571 were modelled by the experimentally validated FE model. The FE model and wave excitation  
572 frequency are the same as those used as Section 5.2. FIG. 14 shows a schematic diagram of  
573 the through thickness locations of the damage. The dimensions of the damage are  
574  $0.2\lambda \times 2\lambda \times 0.1d$  ( $l \times w \times t$ ). The through thickness locations of the damage is represented by the  
575 lower surface of the damage and described by  $z$  axis. Therefore, the locations of the damage  
576 were moved from upper surface  $z = -1$  mm to  $z = -10$  mm with a step of  $0.1d$ . It is noteworthy  
577 that when  $z = -1$  mm and  $z = -10$  mm, the damage is a surface notch located at the upper and  
578 the lower surface of the timber, respectively.

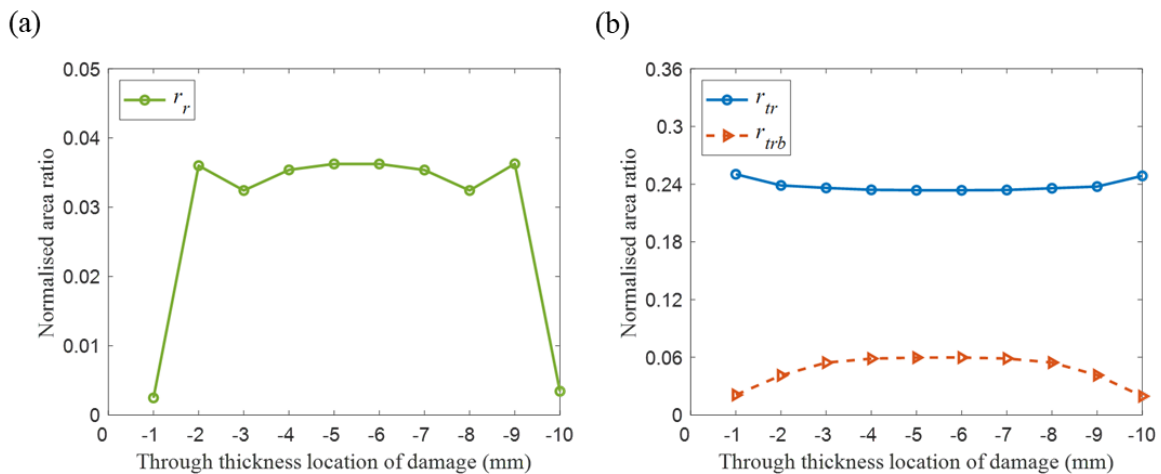
579



580

581 FIG. 14 Schematic diagram of through thickness location of the internal damage

582



583

584 FIG. 15 Normalized area ratio for internal damage with different through thickness  
585 locations (lower surface of the damage in  $z$  axis location) for the (a) reflection wave (b)



586 transmission wave (solid line) and forward scattering wave (dashed line) by 35kHz incident  
587 wave  
588

589 The reflection and transmission results are shown in FIG. 15(a) and (b).  $r_r$ ,  $r_{tr}$  and  $r_{trb}$   
590 show symmetric patterns with respect to the mid-plane of the timber. This is because of the  
591 symmetric out-of-plane displacement mode shape of  $A_0$ . The changes of the damage location  
592 can change the bending stiffness of the timber. Therefore, the response of reflection wave  $r_r$   
593 changes correspondingly. According to FIG. 15(a),  $r_r$  reached the maximum amplitude of at  
594 the mid-plane of the timber at  $z = -5$  mm and  $z = -6$  mm, while  $r_r$  has the minimum amplitude  
595 at the damage is a surface notch at  $z = -1$  mm and  $z = -10$  mm. It is also noteworthy that when  
596 the damage is a surface notch,  $z = -1$  mm and  $z = -10$  mm, the amplitude of  $r_r$  is much smaller  
597 than that is located internally. Therefore, based on the amplitude of  $r_r$ , it is concluded that  $A_0$   
598 GW is more sensitive to the damage located internally than the damage located at the surface  
599 i.e. surface notch at 35 kHz. Furthermore, the fluctuation of  $r_r$  also indicated that even if the  
600 damage size is the same, the reduction in timber stiffness can vary for the different through  
601 thickness locations of the damage.  
602

## 603 **6. Limitations of the proposed study**

604 Despite the reflection and transmission area ratios measured from the experimental internal  
605 damages display a simple monotonical pattern, the application of GW for practical application  
606 in timber damage detection can be more complex. The limitations of the proposed methods are  
607 discussed in this section.

608 The first challenge is the multimodal nature of GW. As shown in FIG. 3, the cut-off  
609 frequency for a 10mm red oak timber is 59kHz. To minimize the complexity in wave analysis  
610 of higher orders and multi-modes, the excitation frequency is chosen below the cut-off  
611 frequency. In real-cases, timber might have a much larger thickness than the specimen in this  
612 study. This results a smaller cut-off frequency and therefore the excitation frequency needs to  
613 be reduced. For structural timber with large thicknesses, a different GW type such as surface  
614 wave, longitudinal wave might be considered as the excitation mode. Moreover, the generation  
615 of internal damage and the measurements for reflection and transmission wave are performed  
616 on a selected region of the specimen in this study. The wood grain of the region is selected to  
617 be smooth, and hence, no natural defects or obvious cracks are presented. When the size of

618 timber is much larger, it is impossible to avoid the existence of timber natural defects.  
619 Therefore, a baseline state of the sample with natural defects is required to extract the damage-  
620 related information, which can become a more complex analysis. To account for the high-  
621 variability in damage conditions and timber properties, the GW technique can be combined  
622 with the data-driven approaches such as machined learning in future research, which paves the  
623 way for real-cases online monitoring.

624

## 625 7. Conclusion

626 This paper has provided a comprehensive wave scattering study from the internal damages for  
627 a structural timber using GW. The study helps to improve the in-situ timber damage assessment  
628 using GW. Different sizes of internal damage have been created experimentally on a structural  
629 timber member using a rotary tool. The reflection and transmission ratios of wave have been  
630 measured before and after the internal damages. The experimental internal damages have been  
631 modelled by the 3D semi-stepped damage in 3D FE simulation. A good agreement has been  
632 achieved between the experimental and the numerical results. In summary, both  $r_r$  and  $r_{tr}$   
633 obtained from the experiment results have simple trend, in which  $r_r$  increases with the size of  
634 the damage while  $r_{tr}$  has the opposite trend. Therefore, it can be used for identifying the  
635 internal damage size in the experiment. To simplify the level of anisotropy, timber properties  
636 have been modelled as transversely isotropic for simulating the wave propagation on the timber,  
637 and have proven to be accurate in the FE simulation. It has been shown that the energy of the  
638 wave concentrates along the fibre direction and the low frequency incident  $A_0$  GW has low  
639 attenuation when propagating along the fibre direction.

640 The experimentally validated FE model was used to study the reflection and transmission  
641 effects of damages with different  $l$ ,  $w$  and  $t$ . In general,  $r_r$  presents an intuitive increasing  
642 pattern compared to  $r_{tr}$  and  $r_{trb}$  and therefore more suitable to be used for identifying the  
643 increasing size of the internal damage. However, the amplitude of  $r_r$  is smaller than the  $r_{tr}$  in  
644 most of the cases. It has also been observed that when varying  $l$  of the damage, two reflection  
645 waves from start and far end of the damage can either interfere constructively, or destructively  
646 due to different arrival times of the two reflection waves. Therefore, a slight fluctuation has  
647 been observed in  $r_r$  for varying the damage  $l$ . In addition, the reflection and transmission  
648 effects of damage with different though thickness locations have been studied. It has been  
649 found that for the same size of damage, when the location of the damage is at the surface, i.e.,

650 a surface notch, the amplitude of  $r_r$  is much smaller than those cases when the damage is  
651 located internally.

652 In summary, this study has confirmed the robustness and sensitivity in detecting  
653 conspicuous damages in timber using  $A_0$  GW. The findings in this study have provided  
654 improved insights into GW interaction with internal damage in structural timber, revealing the  
655 potentials for the application of GW in timber damage detection.

656

## 657 **8. Acknowledgement**

658 This work was funded by the Australian Research Council (ARC) under the grant number  
659 DP210103307. The authors are grateful for this support.

660

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## 778 10. Appendix A

779 Coefficients of Eq. (6) for wave propagates in a transversely isotropic plate

$$780 B_1 = [C_{11}C_{33}C_{44} - C_{13}^2C_{44} - 2C_{13}C_{44}C_{55} + C_{33}C_{55}^2 - (C_{33}C_{44} + C_{33}C_{55}$$

$$781 + C_{44}C_{55}) \rho c^2]/(C_{33}C_{44}C_{55}) \quad (A1)$$

$$\begin{aligned}
782 \quad B_2 &= [C_{11}C_{33}C_{66} + C_{11}C_{44}C_{55} + C_{13}^2C_{55} - 2C_{13}C_{55}^2 \\
783 \quad &\quad - (C_{33}C_{11} + C_{11}C_{44} - C_{13}^2 - 2C_{13}C_{55} + C_{33}C_{55} + C_{55}C_{44} + C_{55}^2)\rho c^2 \\
784 \quad &\quad + (C_{33} + C_{44} + C_{55})\rho^2 c^4]/(C_{33}C_{44}C_{55}) \quad (A2)
\end{aligned}$$

$$785 \quad B_3 = [C_{11}C_{55}^2 - (2C_{11}C_{55} + C_{55}^2)\rho c^2 + (C_{11} + 2C_{55})\rho^2 c^4 - \rho^3 c^6]/(C_{33}C_{44}C_{55}) \quad (A3)$$

786

787 Solutions of Eq. (6) for wave propagates in a transversely isotropic plate

$$788 \quad \alpha_{1,3} = \frac{-B \pm [B^2 - 4AC]^{\frac{1}{2}}}{2A}, \alpha_2 = -\alpha_1, \alpha_4 = -\alpha_3, \alpha_5 = -\alpha_6 = [(\rho c^2 - C_{66})/C_{44}]^{1/2} \quad (A4)$$

789 where

$$\begin{aligned}
&A = C_{33}C_{55} \\
790 \quad B &= (C_{11} - \rho c^2)C_{33} - (C_{55} - \rho c^2)C_{55} - (C_{13} + C_{55})^2 \\
&C = (C_{11} - \rho c^2)(C_{55} - \rho c^2)
\end{aligned} \quad (A5)$$