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Paul R. Medwell and Bassam B. Dally

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Full-length article:

Effect of fuel composition on jet flames in a heated and diluted oxidant stream

Paul R. Medwell^{*}, Bassam B. Dally

School of Mechanical Engineering, The University of Adelaide, S.A. 5005 Australia

Abstract

The role of hydrogen addition on the structure of the Moderate or Intense Low oxygen Dilution (MILD) combustion regime is examined using a combination of experimental techniques and laminar flame calculations. Laser diagnostic imaging is used to simultaneously reveal the in-situ distribution of the hydroxyl radical (OH), formaldehyde (H₂CO), and temperature using the Jet in Hot Coflow (JHC) burner. The fuels considered are natural gas, ethylene, and LPG (each diluted with hydrogen 1:1 by volume). Hydrogen addition to the primary fuel was found necessary to stabilise the flames. Further to the role of hydrogen in the stabilisation of the flames, hydrogen addition also leads to the reaction zone exhibiting similar structure for different primary fuel types. The independence of the reaction zone structure with hydrogen addition suggests that a wide variety of fuels may be usable for achieving MILD combustion.

^{*} Corresponding Author. Tel: +61 (0)8 8303 5460; Fax: +61 (0)8 8303 4367
Email address: paul.medwell@adelaide.edu.au (Paul R. Medwell).

1 Introduction

The combustion of fuel in a heated and diluted oxidant stream results in a unique combustion regime referred to as Moderate or Intense Low oxygen Dilution (MILD) combustion [1]. The strong recirculation of exhaust gases back into the reaction zone results in a localised reduction in O_2 level, leading to a distributed reaction zone. The result of MILD combustion is a reduction in pollutant emissions (notably NO_x) and an increase in net radiation flux [2, 3]. Application of MILD combustion has been successfully applied in numerous applications, additionally incorporating the use of biomass fuel [4]. MILD combustion is typified by low (near unity) Damköhler numbers [5], and has the potential for cost effective low NO_x gas turbine power generation [6].

The depleted O_2 oxidant at elevated temperatures, necessary for MILD combustion, is typically realised by the recirculation of hot exhaust gases. Recirculation may be achieved either internally or externally with regard to the combustor. The complex interactions within such a system make it unsuitable for a fundamental study of the reaction zone. Instead, in this study an experimental burner is used to emulate MILD combustion under controlled conditions. The Jet in Hot Coflow (JHC) burner [7] enables a range of combustion parameters to be varied independently, and decouples the flow from the chemical kinetics. The JHC burner has been used for both experiments [7–10] and modelling [11–19]. A similar burner has also been used for combined experiments [20, 21] and modelling [22].

Despite much progress in the application of MILD combustion to practical systems, there remain unresolved issues on the stabilisation, auto-ignition,

25 and structure of the reaction zone near the jet exit under the hot and diluted
26 conditions [10, 23, 24]. In particular, the influence of fuel type on the funda-
27 mental aspects of the combustion remain poorly understood. From a global
28 perspective, MILD combustion features have been shown to be insensitive to
29 fuel type, including liquid fuels [25]. Similarly, in a furnace environment it has
30 been observed that the combustion of light fuel oil is very similar to natural
31 gas, though differences were observed for heavy fuel oil and coal [26]. Further-
32 more, MILD combustion is particularly well-suited to using low calorific value
33 (LCV) fuels [27]. Notably, the operation of a furnace using natural gas and bio-
34 gas (60% CH₄ / 40% CO₂) has shown similar performance and emissions [28],
35 as shown elsewhere [29]. MILD combustion has been demonstrated for both
36 LCV fuels and industrial waste in the presence of hydrogen [30], with coke
37 oven gas [31, 32], and in gas turbine combustors with LCV fuels [33]. There is
38 also a move toward using coal with MILD combustion [34, 35]. Nonetheless,
39 detailed fundamental understanding of MILD combustion of non-conventional
40 fuels has yet to be widely achieved [31]. While other studies have used both
41 methane and propane for MILD combustion [36], the present paper aims to
42 provide a dedicated detailed fundamental-level investigation on the effect of
43 fuel composition.

44 Hydrogen has been proposed as a “clean fuel” alternative for the future, in par-
45 ticular for use as a supplemental fuel additive [37, 38]. Whilst the use of pure
46 hydrogen eliminates CO₂ emissions, it requires a redesign of existing burners
47 [18]. Avoiding the need for purity, hydrogen may also be used for enrichment,
48 which can be obtained using methods (such as gasification of biomass) that
49 have both economical and environmental benefits [38]. The addition of hy-
50 drogen to other fuels is of particular interest to MILD combustion as it can

51 be well-controlled, improves the range of operating conditions and is also im-
52 portant for the incorporation of low calorific fuels [30]. Furthermore, although
53 MILD combustion is effective at eliminating soot, the addition of hydrogen
54 will further increase the ability to use fuels that have sooting tendencies [31].

55 Hydrogen is commonly used as a fuel enhancer to improve system reactiv-
56 ity, such as to promote ignition of low calorific value fuels [30]. The role of
57 different hydrogen concentration with hydrocarbon (methane) fuel has been
58 studied numerically by Mardani *et al.* [14], and shown to increase the reaction
59 intensity. Hydrogen addition is known to improve laminar flame speed [39],
60 and increase the maximum strain rate before extinction [40]. These features
61 may enable operation over a wider range of conditions [30], but may require
62 the redesign of conventional burners [38]. The addition of hydrogen increases
63 the flame speed, such that flames are more likely to attach to the burner
64 nozzle [38, 41, 42], which may negatively impact the operation of MILD com-
65 bustion burners which require significant mixing of the fuel prior to reaction
66 [31]. As expected, increased hydrogen concentration in the fuel also leads to
67 increased OH concentration in the reaction zone [18, 38]. As a consequence
68 of the reaction occurring closer to the exit, in conjunction with the higher
69 reaction rates, there is less opportunity for the mixing that is required to
70 achieve low-NO_x emissions [43]. Nonetheless, this effect will be dependent on
71 burner configuration (especially burners that rely on entrainment due to high
72 velocity). It has also been reported that NO_x emissions are not sensitive to
73 the hydrogen concentration under MILD conditions [44], though this observa-
74 tion requires accurate understanding of NO formation via the NNH route [45].
75 Further details on NO_x formation are detailed in the literature [46–51]. It is
76 important to note that relatively small quantities of hydrogen are required to

77 have significant changes on the combustion: further increasing the hydrogen
78 concentration does not lead to proportional changes in the behaviour, which
79 is attributed to the role of hydrogen in affecting the chain-branching [31], and
80 highlights the importance of detailed chemistry in modelling such combustion
81 [52]. The influential role of hydrogen addition has been observed to eliminate
82 the region of “no ignition” in the jet stirred flow reactor of Sabia *et al.* [30].

83 In this paper the influence of combustion chemistry on the flame behaviour
84 and reaction zone structure is examined by systematically increasing the fuel
85 complexity under MILD combustion conditions. Temperature, the hydroxyl
86 radical and formaldehyde are measured instantaneously and simultaneously
87 using planar laser imaging techniques to reveal details of the structure of the
88 reaction zone. The hydroxyl radical (OH) is used as a flame marker while the
89 formaldehyde (H₂CO) intermediate species is predominant at low tempera-
90 tures typical of those found in MILD combustion. The product of [OH] and
91 [H₂CO] has also been suggested as an indicator of the formyl (HCO) radical,
92 which is closely related to the heat release rate [53]. Three different fuels are
93 considered, namely; natural gas (predominately methane), ethylene, and LPG
94 (predominately propane). The difference in the chemical path for these fuel
95 mixtures provides a way of assessing the sensitivity of the MILD combustion
96 regime to fuel type (for gaseous hydrocarbon fuels). Each primary fuel is di-
97 luted with hydrogen (H₂) in an equal volumetric ratio to reduce the levels
98 of soot and to stabilise the flame. The role of hydrogen addition is addition-
99 ally investigated through laminar flame calculations. For each fuel type in
100 the jet, differences in the fluid properties necessitates a change in the veloc-
101 ity to maintain the Reynolds number constant (at $Re_{jet}=10,000$) in order to
102 maintain similar turbulence levels.

103 2 Experimental Details

104 The MILD combustion burner used in this study is the jet in hot coflow
105 (JHC) burner used previously [8, 9], and shown in Figure 1. It consists of
106 a central insulated fuel jet ($\varnothing 4.6\text{mm}$) within an annular coflow ($\varnothing 82\text{mm}$) of
107 hot exhaust products from a premixed secondary burner mounted upstream
108 of the jet exit plane. The coflow O_2 level is either 3% or 9% (volumetric), with
109 an exit temperature of 1100K. Various hydrocarbon fuels are used in the jet;
110 natural gas (NG), ethylene (C_2H_4) and liquefied petroleum gas (LPG). Each
111 of these primary fuels is diluted with hydrogen (H_2) in an equal volumetric
112 ratio. The addition of H_2 is found to be necessary to stabilise the flames.
113 Hydrogen addition also lowers the level of soot, which is advantageous for
114 laser measurements. Addition of H_2 also has implications for the potential use
115 of hydrogen as a supplemental fuel additive. Noteworthy is that although the
116 fuel stream is 50% H_2 by volume, it is less than 11% on a mass basis, and
117 the heat release from the hydrogen is $<25\%$. The jet Reynolds number for the
118 experimental data presented in this paper is 10,000.

119 Laser induced fluorescence (LIF) is used to image OH and H_2CO , and tem-
120 perature is inferred from Rayleigh scattering measurements. The laser pulses
121 are fired sequentially to reduce interferences on the other systems, with the
122 entire sequence occurring in 300ns to ensure the flow field is effectively frozen
123 with respect to the fluid time scales. The in-plane resolution of all three ICCD
124 cameras is $160\mu\text{m}$, after spatial matching. The laser sheet heights were all
125 $\sim 12\text{mm}$, of which the central 8mm portion is presented herein. All images
126 are corrected for laser power and profile variations shot-to-shot based on the
127 signal from a laminar slot burner. Description of the experimental details is

128 described in-depth in a previous publication by the authors [8].

129 **3 Laminar Flame Calculations**

130 Laminar flame calculations have been performed to extend the study beyond
131 the available measurements, and shed more light on the structure of the reac-
132 tion zone and molecular transport. The OPPDIF routine of the Chemkin pack-
133 age is used to compute temperature and species concentration for opposed-
134 flow diffusion flames. The opposed-flow laminar diffusion flame configuration
135 represents a one-dimensional flame, analogous to the traverse across the well-
136 defined reaction zone from fuel to oxidant. By increasing the velocity of the
137 flow from two facing nozzles, the strain rate imposed on the flame front can
138 be varied. The strain rate quoted throughout most of this paper is the average
139 normal strain rate reported in the OPPDIF post-processor output. The use
140 of a laminar, one-dimensional, configuration provided by the OPPDIF mod-
141 els is a well established methodology to enable the role of strain alone to be
142 de-coupled from the more complex turbulent interactions that are observed
143 experimentally [10]. Furthermore, it has previously been demonstrated that
144 one-dimensional laminar diffusion flame calculations give excellent agreement
145 with detailed single-point measurements in the JHC burner [7]. Other studies
146 on the fundamental aspects of MILD combustion also have used this configu-
147 ration [54, 55].

148 For all calculations the GRI-Mech 3.0 mechanism is used. Previous stud-
149 ies have shown the GRI mechanism to provide agreement with experimental
150 measurements of the JHC burner used in this study under similar conditions
151 [7, 12, 56]. Both thermal diffusion and multi-component diffusion models are

152 used for the calculations.

153 Since the coflow oxidant stream consists of combustion products (H₂O and
154 CO₂), the standard definition of mixture fraction is not appropriately defined
155 for calculations based on the mass fraction of H & C (hydrogen & carbon)
156 atoms. A normalised mixture fraction, $\xi^* = (\xi - \xi_{oxi}) / (\xi_{fuel} - \xi_{oxi})$ is used
157 instead, where ξ_{fuel} & ξ_{oxi} refer to the standard definition of mixture fraction
158 at the fuel and oxidant stream boundaries, respectively.

159 4 Results

160 4.1 Visual Observations

161 Figure 2 shows photographs of the flames presented in this paper. The hori-
162 zontal dashed lines at 35 & 125mm downstream of the jet exit plane indicate
163 the locations of the laser diagnostic measurements. Apparent from Figure 2 is
164 that the natural gas flames show significantly less soot than the ethylene and
165 LPG flames. While soot is seen at both coflow O₂ levels for the ethylene and
166 LPG flames, at the 3% O₂ level soot does not appear until around ~200mm
167 downstream, whereas for the 9% O₂ case soot appears much closer to the jet
168 exit (~100mm). For either fuel type, at the higher (9%) O₂ level, close to
169 the jet exit the flame luminosity is significantly greater as compared to the
170 lower (3%) O₂ case. The low luminosity of the 3% O₂ flames almost makes
171 them appear invisible for the first ~100mm. While not clearly apparent from
172 the photographs, a faint reaction is indeed apparent in this region. Further
173 downstream, once the effects of the coflow are diminished by the entrainment
174 of surrounding air (*viz.* ~100mm), soot does begin to appear for either coflow

175 O₂ level. The presence of soot around the 125mm downstream location in the
176 9% O₂ flames could potentially interfere with laser diagnostic measurements,
177 and so data is collected at this location for the 3% O₂ flames only.

178 4.2 Typical Features

179 Figure 3 presents typical image triplets of OH, H₂CO and temperature for
180 each flame condition (three fuel types, each at two coflow O₂ levels). The
181 measurements are centred at 35mm downstream of the jet exit plane. The
182 corresponding size of each image is 8mm in height and 35mm wide. The jet
183 centreline is marked by the vertical dashed line. These images are for a jet
184 Reynolds number of 10,000 and are typical of other Reynolds number flames
185 as well. Despite the flow being nominally turbulent, the majority of the images
186 show no sign of large-scale turbulent structure.

187 In each of the images presented in Figure 3 the OH appears as an unconvoluted
188 layer which is quite uniform in intensity along the length of the sheet. For each
189 fuel type, the OH concentration is considerably less at 3% O₂ as compared to
190 the 9% O₂ case. The suppression of OH with a reduction in O₂ level is con-
191 sistent with previous work (*e.g.* [8, 57]) and is directly related to the reduced
192 temperature of the reaction zone. At either O₂ level the OH concentration
193 does not significantly vary with the fuel type.

194 The temperature in the coflow is seen to be uniform. With the 9% O₂ coflow
195 the temperature is seen to increase in the region corresponding to OH. For all
196 of the 3% O₂ cases there is no obvious sign of a temperature increase across
197 the reaction zone, although a reaction is clearly taking place as identified by

198 the presence of OH. Similar observations regarding the very low, almost in-
199 distinguishable temperature rise across the reaction zone has been seen in a
200 MILD combustion furnace [58]. The low measured temperatures are believed
201 to be genuine, and not because of interferences or problems with the Rayleigh
202 to temperature conversion process. Noteworthy is that the laser-based mea-
203 surements of the jet and coflow temperatures agree well with those expected,
204 and were also confirmed with thermocouple measurements.

205 The H₂CO concentration varies with both the O₂ level and even more dra-
206 matically with the type of fuel. Most notably, the 3% O₂ C₂H₄/H₂ flame has
207 significantly higher H₂CO than any other flame. In all cases the H₂CO appears
208 quite uniformly distributed and always exists on the fuel-rich side of the OH
209 layer. The broad radial distribution of H₂CO is also seen in strained laminar
210 flame calculations [9].

211 4.3 Radial Profiles

212 Figure 4 shows the mean and RMS radial profiles of OH, H₂CO, and tem-
213 perature for both 3% and 9% O₂ for the various fuel types, and at an axial
214 location 35mm above the jet exit plane. Each plot is generated only from the
215 central 3mm strip of the images, and not from the entire sheet height, so as to
216 avoid potentially over-corrected values towards the edge of the images where
217 the low laser energy possibly makes sheet corrections less reliable.

218 The instantaneous images presented (Figure 3) suggest that each of the vari-
219 ous fuel types have a very similar structure. This is also seen in the radial plots
220 of Figure 4. At the 3% O₂ coflow the mean OH profiles seem quite coincident,

221 with each fuel having a similar peak mean value and similar spatial shape.
222 These observations are consistent with the similarity of global features ob-
223 served in a MILD combustion reactor [25]. Nonetheless, in both the mean and
224 RMS, there is a slight shift of the OH profiles inward towards the centreline as
225 the fuel complexity is increased. This radial shift corresponds to a drop in jet
226 velocity required to maintain the jet Reynolds number. At the 9% O₂ coflow
227 the radial shift of the OH peak with the fuel type is more noticeable. Also
228 more noticeable for the 9% case is a variation in the mean OH peak, although
229 the changes are still relatively small. Worth noting is that despite an almost
230 three-fold difference in the jet exit velocity for the various fuels, because the
231 Reynolds number is constant in all cases, the OH RMS is comparable for either
232 O₂ level.

233 At both coflow O₂ levels, in the H₂CO profiles of Figure 4 a very significant
234 increase is noted for the C₂H₄/H₂ flame. The mean H₂CO is distributed widely
235 across the radial profiles, but does show evidence of a dip along the jet cen-
236 treline. The broad radial distribution of H₂CO has already been noted in the
237 instantaneous images, and is consistent with strained laminar flame calcula-
238 tions. The H₂CO levels in the C₂H₄/H₂ flame are significantly higher with the
239 3% O₂ coflow as compared to 9% O₂, but for the other fuels (natural gas/H₂
240 & LPG/H₂) the H₂CO is similar at either O₂ level.

241 As was noted in the instantaneous images, the temperature rise across the re-
242 action zone in the 3% O₂ coflow is barely discernible. For the 3% O₂ LPG/H₂
243 flame the temperature rise is not resolved in the mean profile. The lack of tem-
244 perature rise has already been discussed in the typical instantaneous images.
245 It is also important to note that the fluctuations in the radial location cause
246 the peak in the mean radial profile to be significantly lower than instantana-

247 neous temperature peak. The low temperature rise in the mean radial profiles
248 is consistent with previous single-point data measured in the JHC burner un-
249 der MILD conditions [7]. At the 9% O₂ coflow, again the LPG/H₂ flame shows
250 the lowest reaction zone temperature. The trends relating to the location of
251 the peak temperature follows the same trend as seen in the OH profiles, *viz.*
252 the peak shifts towards the centreline as the fuel complexity increases.

253 4.4 Laminar Flame Calculations

254 4.4.1 3% O₂ oxidant

255 The trends noted in the flame measurements can be compared to strained
256 laminar flame calculations. It should be noted that due to the turbulent na-
257 ture of the experimental measurements, the fluctuations in the reaction zone
258 location will cause the mean OH and temperature measurements to be lower
259 than those from laminar flame calculations. Figure 5 shows selected species
260 concentrations in mixture fraction space for a strain rate of $\sim 100 \text{ s}^{-1}$. The
261 temperature and OH, H₂CO & HCO mole fractions were obtained from OP-
262 PDIF calculations of the Chemkin package using GRI-Mech 3.0 mechanism.
263 Previous studies have shown that the reaction structure is relatively insensi-
264 tive to strain under MILD conditions, and that a strain rate of this order gives
265 the best agreement with experimental measurements [7, 8]. A normalised mix-
266 ture fraction (ξ^*) is used instead of the standard definition obtained from the
267 calculations because of the non-standard oxidant stream composition [9]. The
268 normalised mixture fraction ensures that ξ^* ranges from zero in the oxidant
269 stream to one in the fuel stream. To observe features more clearly, the mixture
270 fraction is only shown to 0.25, beyond which there are no features of interest.

271 Consistent with the measurements, the OH profiles in Figure 5 for the various
272 fuels virtually overlap, both in location and peak value. There are some minor
273 differences in the width of the OH profiles in Figure 5, but when normalised,
274 the FWHM (full-width half-maximum) width varies by only around $\sim 5\%$.
275 Similarly, the temperature profiles in Figure 5 across the reaction zone are
276 also very similar, and virtually overlap for the different fuel types. At higher
277 mixture fractions, the temperature for the CH_4/H_2 flame does roll-off a little
278 faster than the others, but the differences are relatively minor. The general
279 observations regarding OH and temperature confirm the experimental result
280 that the fuel type has only minor effects on the reaction zone structure.

281 While the OH and temperature do not seem to vary significantly with the type
282 of fuel, from Figure 5, H_2CO does show very different behaviour depending
283 on the fuel type. In all cases, H_2CO is found almost over the entire range of
284 mixture fraction, extending well beyond the presented range. This explains the
285 very wide radial distribution of H_2CO in the images and reinforces that PAH
286 or Raman interference is not responsible for the distribution seen in the mea-
287 surements. It is therefore more appropriate to make comparisons away from
288 the stoichiometric conditions. It is only for the CH_4/H_2 flame that the H_2CO
289 peak is located near the stoichiometric mixture fraction. Both the $\text{C}_2\text{H}_4/\text{H}_2$
290 and $\text{C}_3\text{H}_8/\text{H}_2$ flames reach a local maximum around stoichiometry, but fur-
291 ther toward the fuel rich side the H_2CO concentration dips slightly and then
292 increases again. Analysis of the H_2CO rate of production indicates that the
293 $\text{C}_2\text{H}_3 + \text{O}_2 \rightleftharpoons \text{HCO} + \text{H}_2\text{CO}$ reaction is primarily responsible for the presence
294 of H_2CO in the fuel rich region. In this region, the concentration of C_2H_3 is
295 higher in the $\text{C}_2\text{H}_4/\text{H}_2$ and $\text{C}_3\text{H}_8/\text{H}_2$ flames, leading to the higher concentra-
296 tion of H_2CO . The localised dip in H_2CO between stoichiometry and the fuel

297 rich region is a result of the increased H_2CO consumption via the $\text{H}+\text{H}_2\text{CO}$
298 $\rightleftharpoons \text{HCO}+\text{H}_2$ reaction.

299 Also plotted in Figure 5 is the HCO mole fraction. As expected, HCO lies in
300 between the OH and H_2CO profiles. The HCO profile does change somewhat
301 with the fuel type. For CH_4/H_2 the HCO is narrower but has a higher peak,
302 whereas $\text{C}_2\text{H}_4/\text{H}_2$ and $\text{C}_3\text{H}_8/\text{H}_2$ are more spread out, but only slightly more
303 so.

304 Noteworthy is that the peak temperature measured is lower than the laminar
305 flame calculations. This is not fully understood, though the effect of radiation
306 which is not accounted for in the calculations may be a contributing factor,
307 along with some uncertainty in resolving the Rayleigh cross-section across
308 the reaction zone. Significantly, the experimental results correctly capture the
309 coflow and jet temperatures.

310 4.4.2 9% O_2 oxidant

311 For the 9% O_2 coflow case, Figure 6 shows the same profiles as Figure 5. The
312 temperature, OH and HCO all seem to have a similar response to the fuel
313 type as already noted for the 3% O_2 case. Consistent with the measurements,
314 the location of the OH peak again does not seem to be highly dependent on
315 the fuel type, and the peak concentration is also relatively constant. At 9%
316 O_2 the differences between the different fuels are somewhat more pronounced,
317 but the general trends and observations are comparable.

318 At 9% O_2 the H_2CO distribution is again seen to be wide, extending well
319 into the fuel rich side. Unlike at 3% O_2 , at 9% O_2 the H_2CO profiles do not

320 change quite as significantly with fuel type, although the differences are still
321 noticeable. The basic profile for each of the fuels is somewhat consistent, with
322 a peak just to the rich side of stoichiometry and a long tail extending further
323 to the fuel rich side. Noteworthy is that unlike the 3% O₂ oxidant, the H₂CO
324 does not increase in the fuel rich region for the 9% O₂ case. This difference is
325 a result of the reduced O₂ concentration in this region at 9% O₂, as compared
326 with 3% O₂, inhibiting H₂CO formation via the $C_2H_3 + O_2 \rightleftharpoons HCO + H_2CO$
327 reaction. The lower O₂ on the fuel rich side of the reaction zone at higher O₂
328 oxidant stream concentration has been discussed previously [10].

329 The stoichiometric mixture fraction, as indicated by the HCO peak, is seen to
330 shift between the different fuels more for 9% O₂ than for 3% O₂. This shift in
331 stoichiometry becomes more noticeable as the stoichiometric mixture fraction
332 becomes larger, but is still relatively small.

333 4.4.3 21% O₂ oxidant

334 To compare the similarity of the reaction zone structure for the different fuels
335 under the hot and diluted O₂ conditions, strained laminar flame calculation
336 results are shown for standard air (21% O₂, 300K) in Figure 7. Apparent is
337 that unlike for the MILD combustion conditions, the fuel type does lead to
338 clear differences in the species profiles. While the OH profiles remain similar, in
339 comparison to the 3% and 9% O₂, 1100K temperature oxidant stream case, the
340 OH profiles are no longer coincident. This suggests that the MILD combustion
341 conditions are fundamentally responsible for bringing about the similarity of
342 the temperature and OH profiles for the different fuels.

343 4.4.4 *Effect of hydrogen addition*

344 The comparison between the three different gaseous hydrocarbon fuels pre-
345 sented thus far has revealed that the fuel type does not have a significant
346 effect on the reaction zone structure under MILD combustion conditions. It
347 is important to highlight that each of the primary fuels were diluted with
348 hydrogen (H_2) in equal volumetric parts. The addition of H_2 was necessary
349 to increase the flame stability to prevent blow-off. Hydrogen also shifts the
350 stoichiometric mixture fraction to the lean side, towards the edge of the shear
351 layer. The H_2 added to the fuel stream seems to influence the kinetics such
352 that the different hydrocarbon fuels show similar characteristics.

353 It was found from experimentation that when H_2 was not added to the jet the
354 flames would typically blow-off. Only the C_2H_4 flame could be sustained, and
355 even then, it appeared lifted. The importance of H_2 addition to the fuel is also
356 reflected in laminar flame calculations. For the experimental 3% O_2 oxidant
357 stream conditions, calculations show no temperature rise across the reaction
358 zone for the pure fuels without H_2 in the fuel stream, except at a very low
359 strain rate. These trends in the calculation for the undiluted fuel types are
360 in agreement with those noted during the experiments with the JHC burner.
361 These results also highlight the importance of H_2 to ignition, and thus the
362 importance of hydrogen-enrichment in the context of practical applications.

363 To examine the structural differences without H_2 addition to the fuel, Fig-
364 ure 8 shows selected species concentrations found from strained laminar flame
365 calculations. The conditions for Figure 8 are identical as for the previously
366 presented Figure 6, but without H_2 added to the fuel. As outlined in the pre-
367 ceding paragraph, comparisons can only be made at the 9% O_2 case, as at the

368 3% O₂ conditions a reaction could not be sustained.

369 Comparison of Figure 8 (no H₂ addition) to Figure 6 (with H₂ addition) reveals
370 major differences in the selected species concentrations. Unlike with H₂ diluted
371 fuels, Figure 8 shows that the OH profiles are dependent on the fuel type.
372 For C₂H₄, the peak OH concentration is 30% higher, and the FWHM (full-
373 width half-maximum, in the normalised mixture fraction space) is 40% wider,
374 than for CH₄ fuel. Significant differences for the various fuel types in the
375 H₂CO and HCO profiles are also far more pronounced than seen with H₂
376 addition. Comparison of the peak concentration in the reaction zone of H₂CO
377 and HCO reveals that both increase by approximately a factor of 3.5 without
378 H₂ addition.

379 The role of H₂ leading to a similar reaction zone structure is important for the
380 use of low calorific value fuels. These results suggest that H₂ not only improves
381 reactivity but also changes the structure of the reaction zone.

382 5 Discussion

383 From the instantaneous images and the radial profiles that have been pre-
384 sented there does not seem to be any significant effect of the fuel type on
385 the structure of the reaction zone. Similar trends are also observed at other
386 measurement locations. The trends obtained from methane and propane fuels
387 have previously been noted as being similar in a heated and diluted oxidant
388 stream in a spectral emission study [59], and are consistent with the similar
389 global behaviour reported previously [25, 26, 28, 29].

390 Of the measured scalars, the H₂CO number density changes the most between

391 the different fuel types considered. Nonetheless, the basic behaviour of the
392 H_2CO is essentially constant between the different fuel cases. Changes in the
393 flow structure are also apparent for the different fuel types. These differences
394 arise because of the different jet velocity for each fuel type (required to main-
395 tain constant Reynolds number). Despite these minor differences, the fuel type
396 does not lead to any major changes in the overall flame characteristics at the
397 measurement locations under MILD conditions.

398 The photographs of the flames (Figure 2) provide supplemental evidence sup-
399 porting the similarity of the different fuel types when in the confines of the hot
400 and diluted coflow. Further downstream, after the effects of the coflow have
401 diminished, each of the flames visually appear significantly different. This is
402 suggestive of the importance of the coflow in establishment of the unique con-
403 ditions which lead to the similarity between the different fuels.

404 Table 1 shows the average of the peak OH values in each of the images for a
405 particular flame. Also included is the standard deviation (as a percentage) of
406 the values. In determining the peak value in each of the images, only the central
407 3mm portion of the image is included to avoid over-corrected values towards
408 the edges of the images where the low laser power makes sheet corrections
409 less reliable. Table 1 reiterates the similarity of the OH levels for the different
410 fuel compositions. At either O_2 level the mean peak OH number density is
411 very similar for each fuel. Consistent with the radial profiles, the similarity is
412 especially noted for the 3% O_2 case.

413 At the 35mm location, where the oxidant composition is well defined and not
414 yet affected by surrounding air entrainment, comparisons of the trends can be
415 made to the strained laminar flame calculations. Table 2 presents the peak

416 temperature, and the maximum number density of OH and H₂CO for each
417 fuel type and at both O₂ levels obtained from the OPPDIF calculations.

418 Comparing Tables 1 & 2 it is seen that calculations support the measurements,
419 in that the OH concentration does not change significantly with the fuel com-
420 position. The experimental peak OH value for the 3% O₂ flames is higher than
421 from the calculations, whereas at 9% O₂ the calculated value is higher than
422 that found from the experiment. The typical differences between the measured
423 value and calculations is approximately 25% for 3% O₂ and ~40% for 9% O₂.
424 Nevertheless, there is good similarity of the trends, and also a similar order of
425 magnitude between the experiments and calculations.

426 Table 2 shows that the H₂CO in the C₂H₄/H₂ flame behaves the opposite
427 to the other two fuels. For C₂H₄/H₂ the H₂CO significantly increases at the
428 lower O₂ case. In contrast, the effect of O₂ on H₂CO is comparatively minor
429 for the other fuels. For C₃H₈/H₂ there is little difference between the two O₂
430 levels. In the CH₄/H₂ flame the trend is reversed and H₂CO slightly increases
431 with O₂ level. The different behaviour of the H₂CO in the C₂H₄/H₂ flames
432 to the other fuels was also apparent in the experimental data presented. Of
433 particular note in the C₂H₄/H₂ flames was that the H₂CO levels in the 3%
434 O₂ flames was much higher than the other cases, which is consistent with the
435 laminar flame calculations (Table 2). At 9% O₂, Table 2 indicates that the
436 C₂H₄/H₂ flame should have lower H₂CO than the other flames, which is not
437 seen in the experimental data. Nevertheless, in general, the trends of H₂CO
438 largely follow those predicted by the flame calculations shown in Table 2.

439 It is important to highlight that each of the primary fuels are diluted with
440 hydrogen (H₂) in equal volumetric parts. The addition of H₂ is necessary to

441 increase the flame stability to prevent blow-off. The H_2 added to the fuel
442 stream seems to influence the kinetics such that the different hydrocarbon
443 fuels show similar characteristics.

444 **6 Conclusions**

445 The comparison between the three different gaseous hydrocarbon fuels in this
446 paper reveals that the fuel type does not have a significant effect on the re-
447 action zone structure under MILD combustion conditions, when mixed with
448 hydrogen. The primary fuels considered were natural gas, ethylene, and LPG
449 (each diluted with H_2 1:1 by volume). Both experimental measurements and
450 strained laminar flame calculations indicate that the reaction zone structure
451 is very similar for the different fuels considered when hydrogen is added to
452 the fuel stream. Hydrogen was found necessary for the experimental flames to
453 stabilise. The OH concentration results were seen to be quite constant. Only
454 minor changes of the OH spatial distribution were noted, and attributable to
455 differences in fuel velocity required to maintain constant Reynolds number.
456 The only significant changes with the fuel type were noted in the H_2CO lev-
457 els, most notably with the C_2H_4/H_2 flame. The trends in the measured H_2CO
458 levels were seen to be consistent with flame calculations. The similarity of the
459 combustion characteristics for the various gaseous hydrocarbon fuels consid-
460 ered suggests that MILD combustion should be readily adapted for different
461 fuel types. The insensitivity to fuel type is potentially a significant advan-
462 tage for the implementation and application of MILD combustion to practical
463 systems.

464 **Acknowledgments**

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466 their assistance with this project. The financial support of The University of
467 Adelaide and the Australian Research Council is gratefully acknowledged.

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Fuel compos- ition	Peak n_{OH} [σ] ($\times 10^{16} \text{cm}^{-3}$)			
	3% O_2		9% O_2	
NG/ H_2	0.56	[11%]	1.28	[11%]
$\text{C}_2\text{H}_4/\text{H}_2$	0.56	[10%]	1.53	[30%]
LPG / H_2	0.54	[9%]	1.33	[8%]

Table 1

Mean and standard deviation (σ , in brackets) of peak OH number density measurements. Central 3mm strip of images used. Axial location 35mm above jet exit.

Fuel compos- ition	Peak temp. (K)		Peak n_{OH} (10^{16}cm^{-3})		Peak $n_{\text{H}_2\text{CO}}$ (10^{14}cm^{-3})	
	3%	9%	3%	9%	3%	9%
	CH_4/H_2	1400	1876	0.40	2.04	1.15
$\text{C}_2\text{H}_4/\text{H}_2$	1403	1856	0.46	2.11	2.59	0.87
$\text{C}_3\text{H}_8/\text{H}_2$	1384	1807	0.43	1.97	1.60	1.45

Table 2

Peak temperature and OH & H_2CO number density (molecules/ cm^3) from strained laminar flame calculations for 3% and 9% O_2 oxidant stream conditions ($a \approx 200\text{s}^{-1}$).

List of Figure Captions

Figure 1: Cross-sectional diagram of jet in hot coflow (JHC) burner.

Figure 2: Photographs of natural gas (NG), ethylene (C_2H_4) & LPG flames, each diluted with hydrogen (1:1 vol/vol) at two coflow O_2 levels. Jet Reynolds number of 10,000. Note the different exposure times (all other camera parameters held constant). Horizontal lines indicate measurement locations (35mm & 125mm downstream of jet exit plane). Photograph height: 300mm.

Figure 3: Selection of instantaneous OH, H_2CO and temperature image triplets of natural gas/ H_2 , C_2H_4/H_2 and LPG/ H_2 flames showing typical features. OH shown in number density (molecules/ cm^3), H_2CO in arbitrary units, temperature in Kelvin. $Re_{jet}=10,000$. Each image $8\times 35mm$. Jet centreline marked with dashed line. Axial location 35mm above jet exit.

Figure 4: Mean and RMS radial profiles of OH, H_2CO and temperature for natural gas/ H_2 , C_2H_4/H_2 and LPG/ H_2 flames. OH shown in number density (molecules/ cm^3), H_2CO in arbitrary units, temperature in Kelvin. $Re_{jet}=10,000$. Central 3mm strip of images used. Axial location 35mm above jet exit.

Figure 5: Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 3% O_2 coflow composition ($a \approx 100s^{-1}$).

Figure 6: Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 9% O_2 coflow composition ($a \approx 100s^{-1}$).

Figure 7: Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for (21% O_2 , 300K) air

($a \approx 100\text{s}^{-1}$).

Figure 8: Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 9% O_2 coflow composition ($a \approx 100\text{s}^{-1}$).

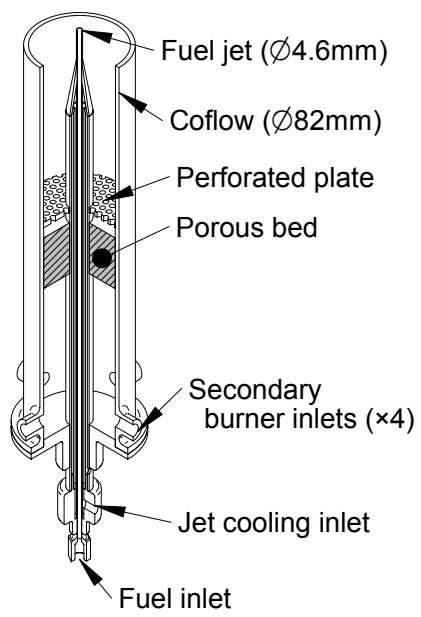


Fig. 1. Cross-sectional diagram of jet in hot coflow (JHC) burner.

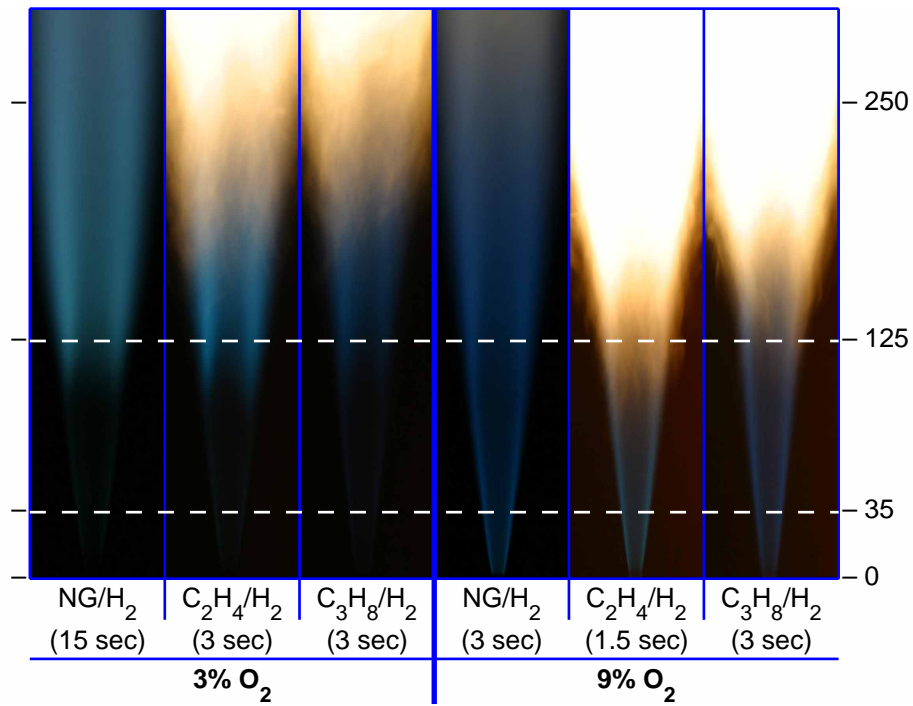


Fig. 2. Photographs of natural gas (NG), ethylene (C₂H₄) & LPG flames, each diluted with hydrogen (1:1 vol/vol) at two coflow O₂ levels. Jet Reynolds number of 10,000. Note the different exposure times (all other camera parameters held constant). Horizontal lines indicate measurement locations (35mm & 125mm downstream of jet exit plane). Photograph height: 300mm.

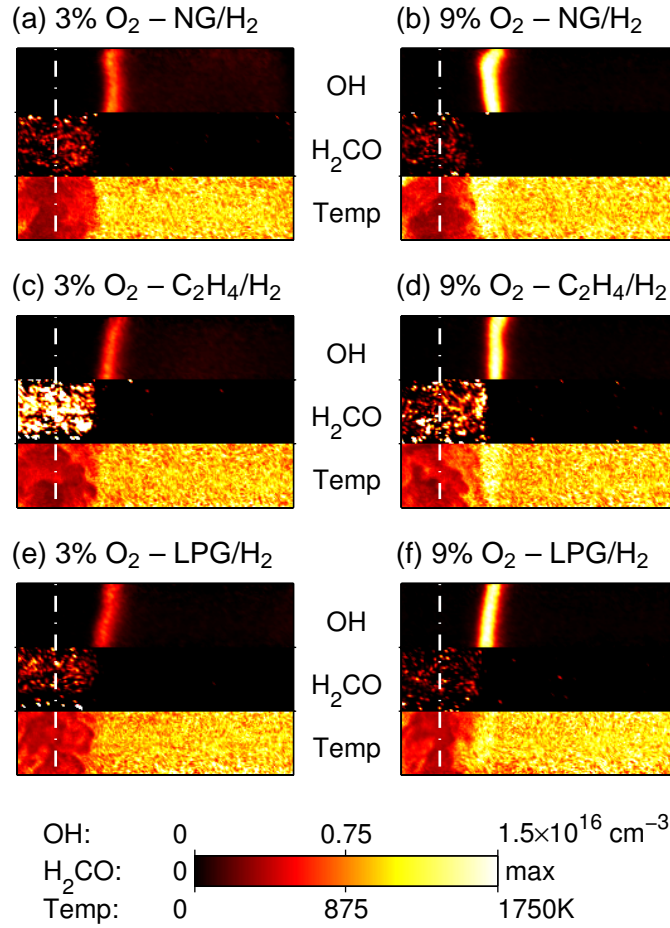


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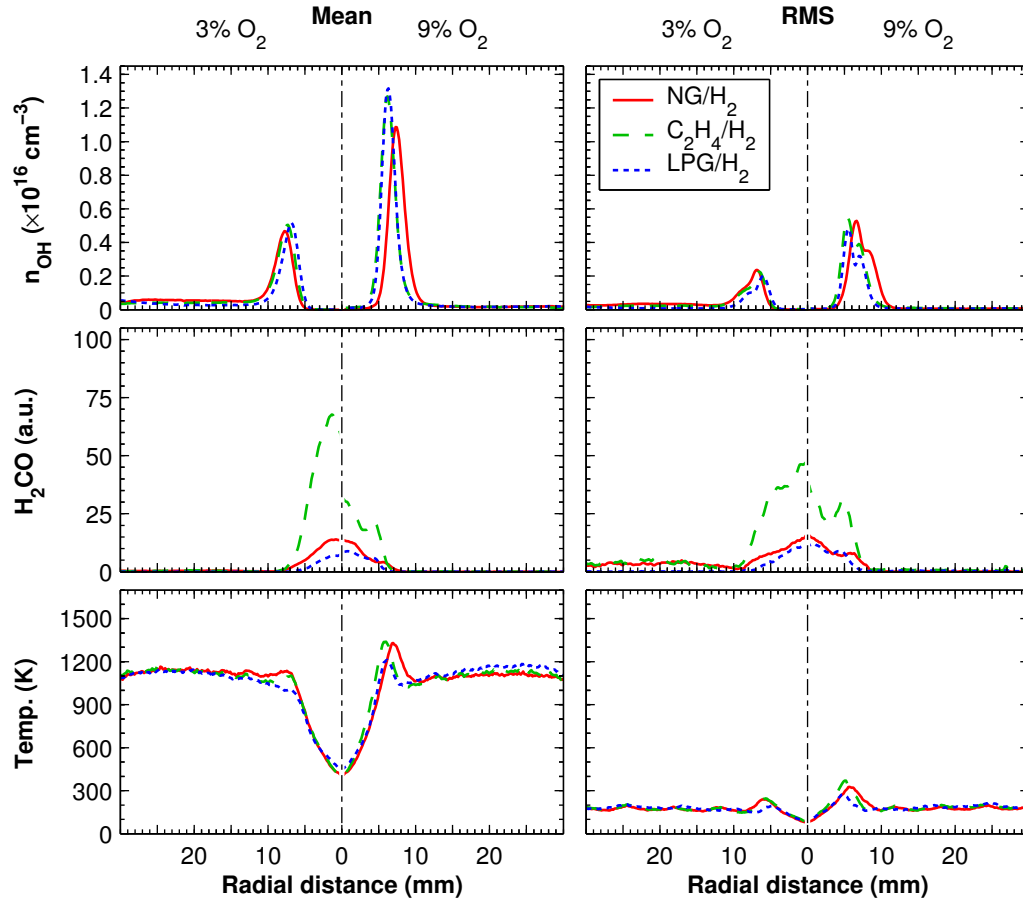


Fig. 4. Mean and RMS radial profiles of OH, H₂CO and temperature for natural gas/H₂, C₂H₄/H₂ and LPG/H₂ flames. OH shown in number density (molecules/cm³), H₂CO in arbitrary units, temperature in Kelvin. $Re_{jet}=10,000$. Central 3mm strip of images used. Axial location 35mm above jet exit.

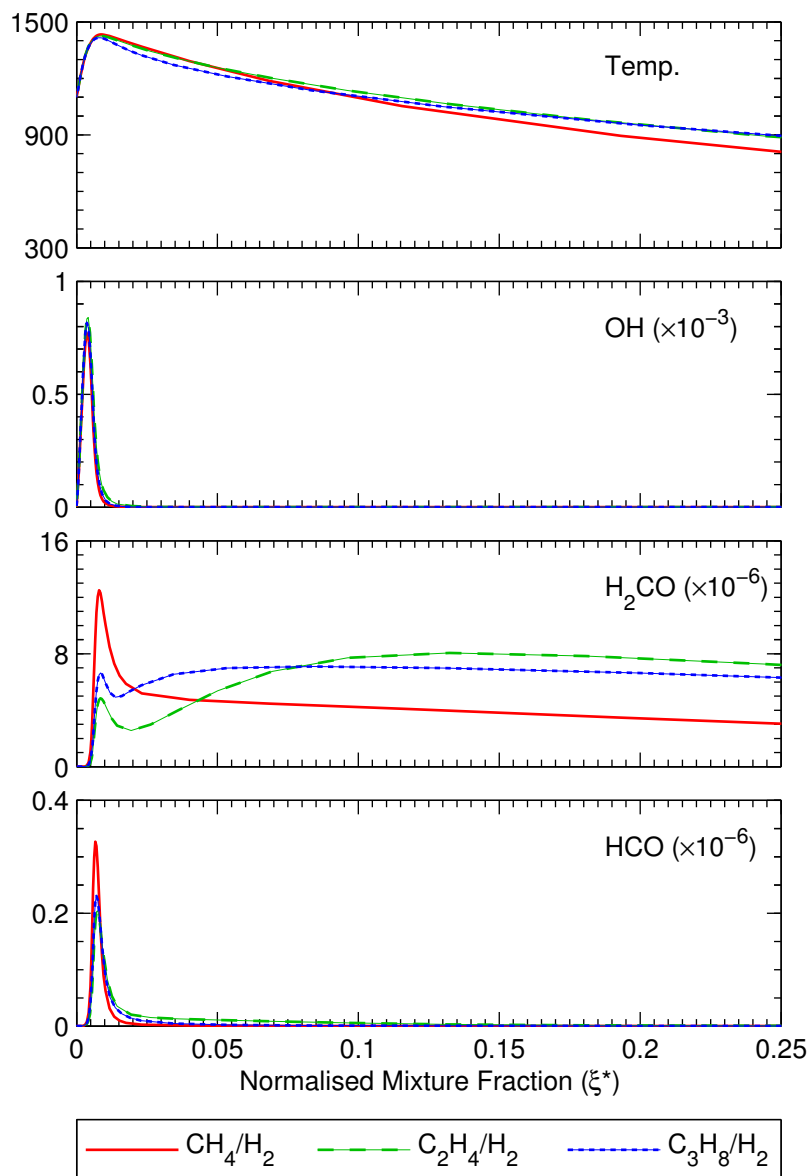


Fig. 5. Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 3% O_2 coflow composition ($a \approx 100\text{s}^{-1}$).

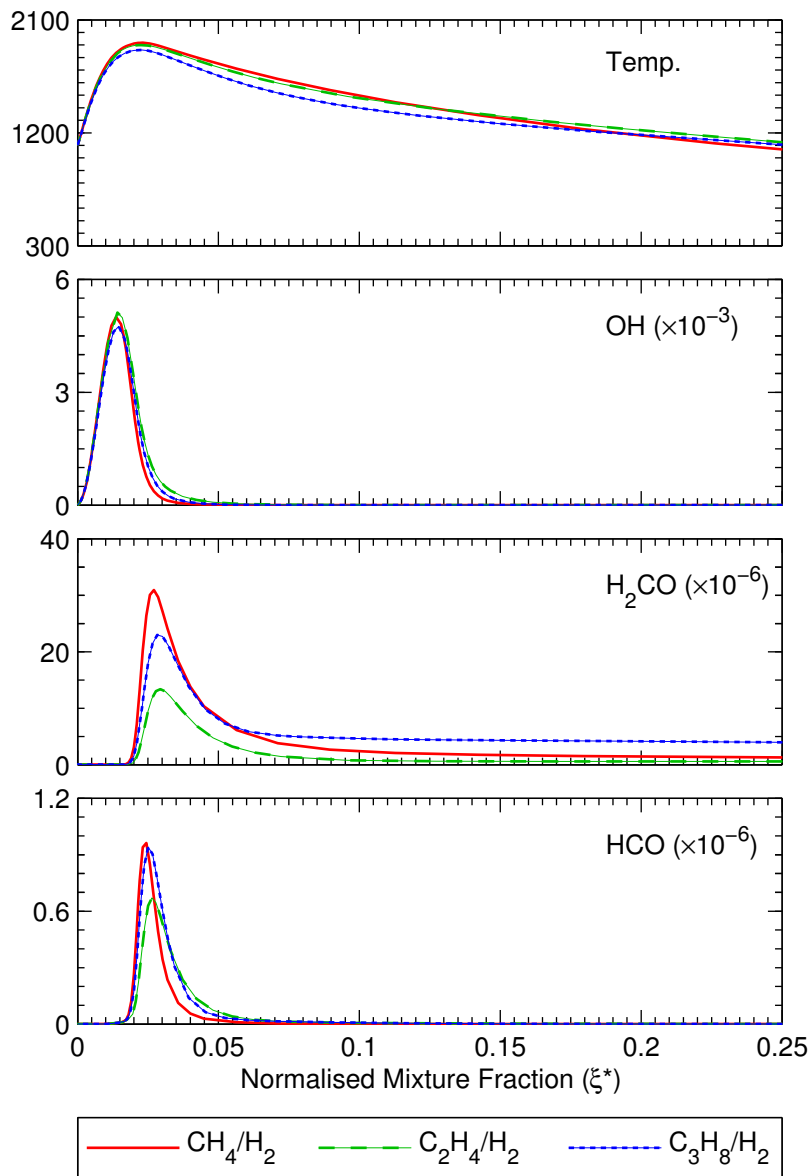


Fig. 6. Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 9% O_2 coflow composition ($a \approx 100s^{-1}$).

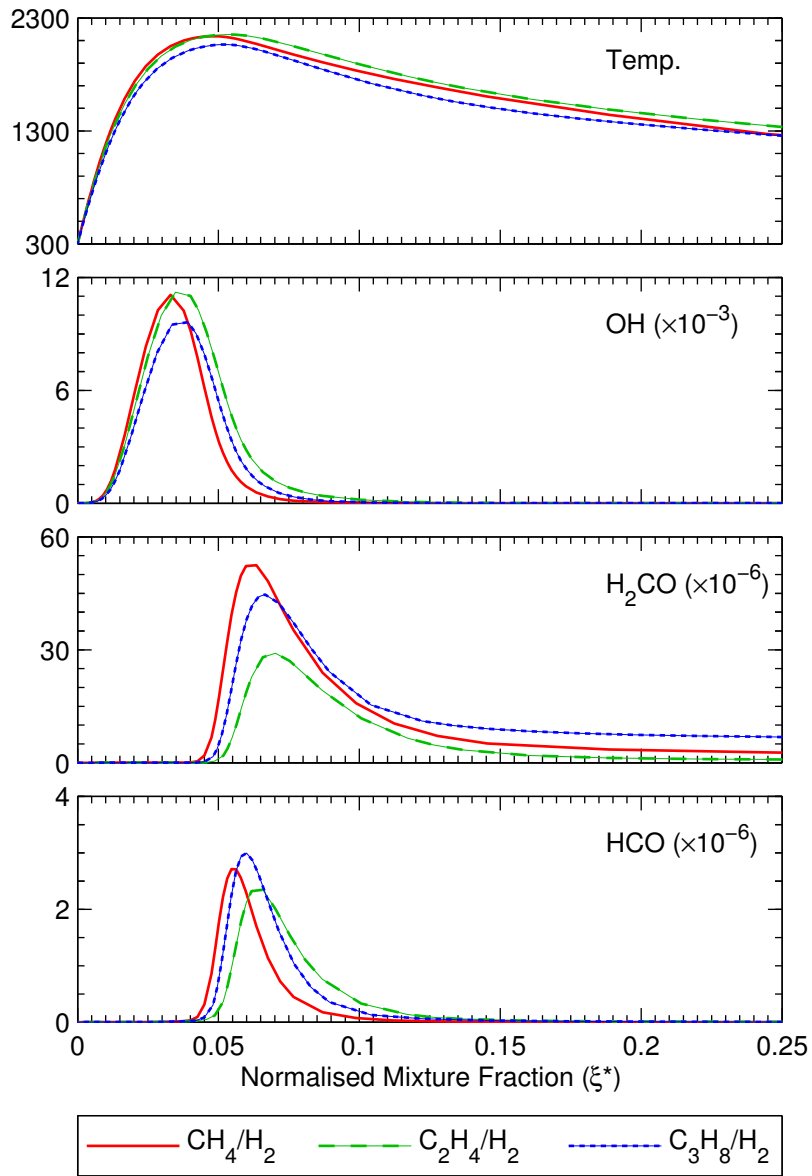


Fig. 7. Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for (21% O₂, 300K) air ($a \approx 100\text{s}^{-1}$).

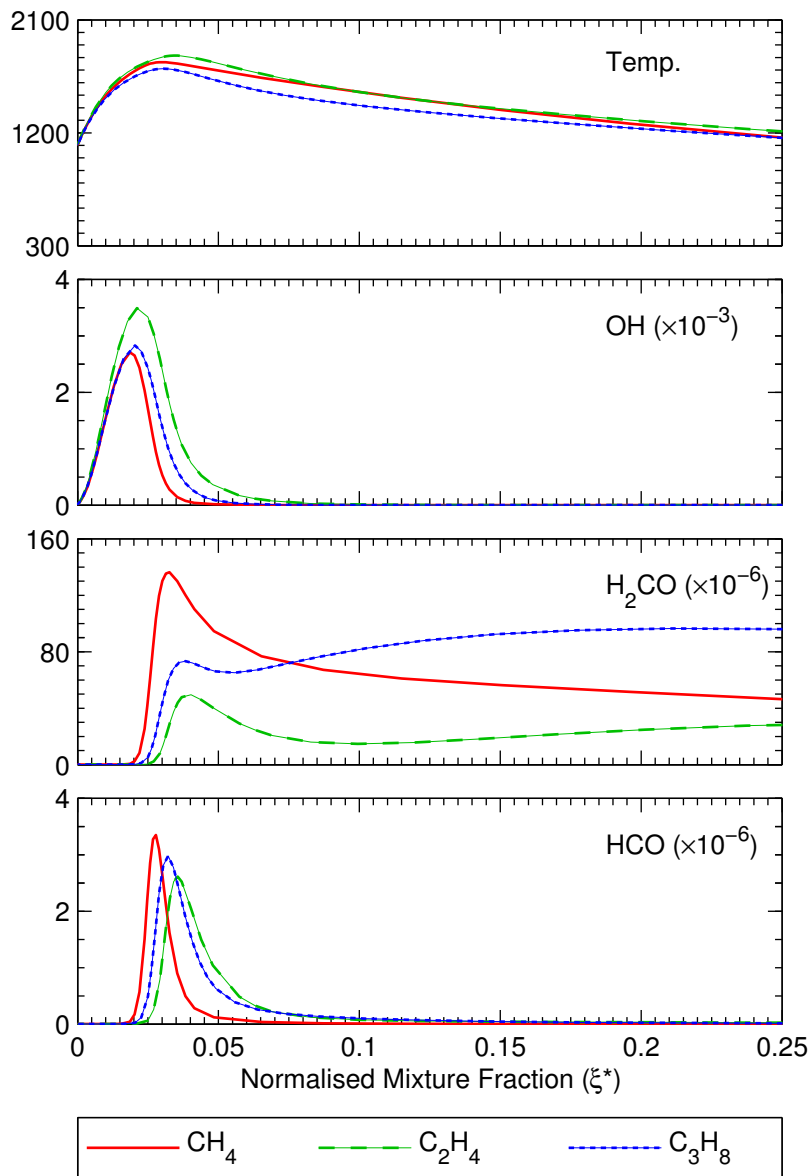


Fig. 8. Temperature and species mole fractions from strained laminar flame calculations in (normalised) mixture fraction space for 9% O_2 coflow composition ($a \approx 100\text{s}^{-1}$).